Laser Materials Processing for NASA's Aerospace Structural Materials

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ABSTRACT

Lasers are useful for performing operations such as joining, machining, built-up freeform fabrication, and surface treatment. Due to the multifunctional nature of a single tool and the variety of materials that can be processed, these attributes are attractive in order to support long-term missions in space. However, current laser technology also has drawbacks for space-based applications. Specifically, size, power efficiency, lack of robustness, and problems processing highly reflective materials are all concerns. With the advent of recent breakthroughs in solid-state laser (e.g., diode-pumped lasers) and fiber optic technologies, the potential to perform multiple processing techniques in space has increased significantly. A review of the historical development of lasers from their infancy to the present will be used to show how these issues may be addressed. The review will also indicate where further development is necessary to realize a laser-based materials processing capability in space. The broad utility of laser beams in synthesizing various classes of engineering materials will be illustrated using state-of-the art processing maps for select lightweight alloys typically found on spacecraft. Both short- and long-term space missions will benefit from the development of a universal laser-based tool with low power consumption, improved process flexibility, compactness (e.g., miniaturization), robustness, and automation for maximum utility with a minimum of human interaction. The potential advantages of using lasers with suitable wavelength and beam properties for future space missions to the moon, Mars and beyond will be discussed.

The laser processing experiments in the present report were performed using a diode pumped, pulsed/continuous wave Nd:YAG laser (50W max average laser power), with a 1064 nm wavelength. The processed materials included Ti-6Al-4V, Al-2219 and Al-2090. For Phase I of this project, the laser process conditions were varied and optimized to see the effects on melt-quenching, cladding/alloying (using the pre-placed powder technique), and cutting. Key parameters such laser power, pulse repetition frequency, process speed and shield gas flow and the observed process characteristics such as plasma formation during laser/material interaction, have been reported for all experimental runs. Preliminary materials characterization of select samples was carried out using various microscopy, diffraction, spectroscopy and microhardness test methods, and reported. Select nitridation results of Ti-6Al-4V using nitrogen assist gas indicated the successful formation of hard titanium nitrides with much higher hardness (2180 kg_f/mm²). A cost-effective and simple powder delivery system has been successfully fabricated for the further experimentation in Phase II.

INTRODUCTION AND LITERATURE BACKGROUND

- The development of a universal, multifunctional tool is essential to support long-term missions in space. In addition to maximized utility, launch costs and the availability of power in space are the primary constraints that will determine whether a tool is sufficiently useful to be included in any mission (with or without a human presence).

 Lasers are useful for performing operations such as surface engineering, built-up freeform fabrication, joining, and machining (Nagarathnam and Taminger, 2001). Additionally, lasers are capable of performing these processes on a wide variety of different materials, including metals, ceramics, polymers, and composites. These attributes are attractive in supporting long-term space missions due to the multifunctional nature of a single tool and the variety of materials that can be processed.
- However, when considering space-based applications, current laser technology also has significant drawbacks, which need to be addressed. Current lasers tend to be large, have low power efficiency, have significant maintenance requirements, and have difficulty with processing highly reflective materials such as aluminum. Currently, since launch costs per pound are extremely high and the available power supplies are quite limited, these drawbacks are not insignificant.
- With the advent of recent breakthroughs in solid-state laser (e.g., diode-pumped Nd:YAG lasers) and fiber optic technologies, the potential to perform multiple processing techniques in space is increasing significantly. Aluminum is a very common material for satellite and spacecraft structures, so techniques need to be developed to be able to effectively process highly reflective materials like aluminum to increase the utility of the laser as a space-based tool. Recent developments in laser technology are potentially enabling for both short- and long-term space missions to explore the limits of innovative processing strategies which combine low power consumption, improved process flexibility, compactness (e.g., miniaturization) and robustness.

Energy Requirements for Laser-Assisted Materials Processing

- Figure 1 shows general regimes for various material processing parameters for both pulsed and continuous wave lasers. Short pulses (ns, ps or fs) with high peak power densities are desirable for laser shock processing, drilling, machining, and ablation applications. In general, longer pulses (ns or ms) or continuous wave lasers are preferred for melting, freeform fabrication, and welding operations, as well as hybrid combinations of processing such as melt-quenching/heat treatment. Laser chemical vapor deposition and laser surface transformation hardening require lower power densities and interaction times as compared to processes involving melting and vaporization.
 - Representative laser processing research efforts and the resulting properties (e.g., grain refinement, less segregation, enhancement of corrosion resistance, an increase in microhardness, fatigue strength improvement) of aluminum and titanium based alloys are summarized in Table 1. These results demonstrate that both CO₂ and Nd:YAG lasers are capable of performing numerous operations on highly reflective, non-ferrous alloys.

Laser Surface Engineering

Materials have been traditionally processed in such a way that their properties were optimized with respect to bulk as well as surface constraint requirements. This is an inherently difficult task as the properties required at the surface (fatigue, wear and corrosion) are often very different from those in bulk form (strength and toughness). The

use of lasers represents a significant development with the potential for optimizing engineered surfaces without compromising the bulk properties.

Laser surface modification has been proven to improve the wear resistance and/or high temperature corrosion resistance in several ferrous and non-ferrous alloy systems. Alternatively, coatings treated by other processing techniques such as plasma spraying or electroplating may be modified with laser beams. Lasers in the infrared regions (~1064 nm in wavelength) have also been used to process materials such as aluminum, titanium and copper alloys. Unlike conventional castings or heat treatment, it is possible to have much higher cooling rates with laser processing. These rapid solidification rates yield superior fine-grained microstructures with metastable crystalline and/or amorphous phases, uniform alloy composition, and extended solubility of alloying elements.

Built-up Freeform Fabrication

Shallow depth of penetration of laser energy into the bulk material can be a benefit, as seen in the surface engineering field. However, to affect bulk material properties with laser processing, techniques originally developed for laser surface cladding or alloying needs to be applied in a built-up process. One built-up freeform fabrication process employs a laser beam to create a small melt pool on the surface of a metallic substrate. Metallic powder is fed into the molten pool in a layer-by-layer process. This results in a fully dense, structural part, built directly from computer-aided design models, without molds, tooling, or machining. The resulting microstructure throughout the bulk is typically much finer than that in conventionally produced materials due to the rapid solidification kinetics associated with the small melt pool. Due to the ability to blend alloy powders, it is possible to produce unique materials that are compositionally tailored and functionally graded in a unitized structure. The process is flexible enough to be able to build near-net shaped parts in complex geometries (which can be net-shape if surface roughness is not an issue). Although not documented in other materials to date, two distinctly different microstructures were obtained in Ti-6Al-4V with different laser settings (Schlienger, 1998). This provides evidence that not only compositions but also microstructures may be controlled and tailored during the fabrication process.

Laser freeform fabrication, using fiber optic delivery, has tremendous potential for revolutionizing the use of lasers in space manufacturing applications because it provides a means by which to build up structures. This process can be modified to use wire feed to eliminate the handling of powder feedstock in a zero or low-gravity atmosphere. Typically, the process is performed in a glove box under inert gas shielding, but the enclosure and gas shielding requirements can be eliminated in the vacuum of space. The built-up freeform fabrication process is computer controlled, and could be fully automated to operate with minimal human intervention. All of these attributes are advantageous for crewed or uncrewed space manufacturing applications.

Laser Welding, Machining, and Other Manufacturing Applications

Lasers are useful in other areas of manufacturing such as welding, micro-machining, solid-state transformation hardening, melt quenching, melt-particle injection, and surface processing. Lasers can also be used on most materials, including ferrous, non-ferrous, ceramic, cermet and composite engineering materials. For example, autogenous laser welding has been successfully performed on numerous ferrous and non-ferrous alloys. Laser welding produces very small heat affected zones and typically requires no post-weld heat treatment. Precision laser micro-machining can be accomplished on a variety of different materials, including metal matrix composites (MMC's). This is significant as MMC's can cause rapid tool wear in traditional machining because of the hard reinforcement phase within the more ductile metal matrix. Nd:YAG lasers have been used to micro-machine SiC reinforced aluminum composites without a large heat affected zone or recast layer (Nagarathnam, et al., 1997). Since lasers are non-contacting, they do not suffer tool wear. Lasers are also capable of machining extremely

delicate details using a small beam diameter and, with fiber optic technology, can perform keyhole drilling. Laser melt-particle injection is used to create in-situ composite coatings by melting only the substrate and injecting hard particles like oxides or carbides during the co-deposition.

The required laser power density, specific energy and interaction time requirements typical for CO₂ laser materials processes are given in Table 2. From Table 2, it becomes obvious that surface melting processes require much higher power densities when compared to solid-state transformation hardening. When using solid-state lasers with shorter wavelengths (1064 nm or lower), the reflectivity problems with aluminum alloys are reduced significantly. The required energy densities and times are correspondingly reduced due to higher absorbtivities, thereby also reducing the input energy requirements.

Based on several experimental findings, typical laser processing maps were developed for surface engineering applications, and are shown for laser surface melt quenching (Figure. 2), laser surface alloying (Figure. 3), laser surface cladding (Figure. 4), laser beam welding (Figure. 5a,b and Appendices A, B, and C) (Nagarathnam and Mazumder, 1996), and laser shock processing (Berthe et al., 1998, Appendix D). These laser-processing maps show the required laser parameters for a variety of ferrous and non-ferrous based alloys for each of the different techniques. Processing maps are useful for defining key laser variables for processing new alloys, and for comparing the range of laser settings used for performing different laser techniques. Most of these techniques can be carried out with the same laser setup by proper choice of fixturing. Addition of feedstock supply and computer-controlled positioning capability can expand a standard laser arrangement to be capable of built-up freeform fabrication. Thus, if a single type of laser can attain the range of settings in these different processing maps, that laser would be extremely versatile in performing many different processing operations on a variety of materials.

Comparison of Laser Types

The theory of light emission dates back to Albert Einstein in 1916. Einstein's theory was brought to reality with Theodore Maiman's engineering use of a ruby laser in 1960. For the past four decades, the CO₂ laser has been the workhorse in various industrial materials processing applications. Over the past decade however, the yttrium aluminum garnet (YAG) solid-state laser has become increasingly more popular for manufacturing applications, due to better absorptivity and the ability to use fiber optics for beam transport.

In pulsed solid-state lasers such as Nd:YAG, Q-switching is commonly used to obtain pulses of various lengths and frequencies (up to 50 MHz). Using a frequency doubling technique, the 1064 nm wavelength can be sent through aligned crystals such as lithium niobate (LBO) to obtain a visible green wavelength (532 nm). The rod geometry of the YAG has been modified into a slab- or face-pumped geometry to generate very high peak powers with reduced thermal lensing and beam distortion. The diode-pumped solid-state laser has a much longer life (several thousands of hours) than that of the krypton arc lamp-pumped lasers, which lasts only few hundred hours. The laser diodes used for pumping have higher brightness, monochromaticity, and more efficient coupling with the lasing medium such as neodymium ions. The compact size of these diodes is also very attractive in reducing the thermal losses in these types of lasers. Some solid-state lasers such as the Ti-sapphire laser are capable of ultra-short pulses (picosecond or femtosecond). Although attractive for diagnostics, micro/nano-machining and ablation in a small scale, this type of laser is less suitable for melting/cladding processes.

A number of alternative lasers with varying wavelengths (Appendix F) and power levels are also available. Both fast axial flow and transverse flow continuous wave gas lasers are available with high power levels up to 100 kW. Among the pulsed lasers, the transverse excitation atmospheric (TEA) laser is popular in attaining high peak power levels of ~1MW, higher pulse energy (10 J) and greater pulse length (2-3 ms). Excimer UV lasers (KrF, XeCl) are pulsed chemical lasers with UV wavelengths, very short pulse lengths, and high peak, but fairly low average power

levels of only a few watts. Copper vapor lasers (532 nm wavelength) exhibit intermediate average and high peak power levels. Chemical oxygen-iodine lasers (COIL, 1315 nm wavelength) have scalable high power levels up to 40 kW. However, chemical and gas lasers are typically bulky in size and require handling chemicals, frequent maintenance, and elaborate cooling systems. These features tend to make them less attractive for processing in space.

Semiconductor lasers are the most compact of all laser types. Semiconductor lasers are typically GaAs, GaAlAs (750-870 nm wavelength), and InGaAs (900-1000 nm wavelength). The power requirements for semiconductor lasers are substantially lower than for other types of lasers, and cooling requirements are minimal. Although these semiconductor (diode) lasers are available at low power levels, they can be used for efficient pumping of solid-state lasers (Appendix E). The desirable wavelengths for optimum energy absorption of reflective alloys include visible (511 nm) to near-infra-red (1064 nm), which are found in diode-pumped solid-state lasers. The wall plug efficiency of lamp-pumped Nd:YAG lasers is only 2%, while diode-pumped solid-state lasers possess about 8% efficiency. CO₂ lasers have a typical efficiency rating of approximately 12%. However, the advent of semiconductor lasers as pumping sources has increased efficiencies up to 30-40%. Although the demonstrated increases in efficiency are encouraging, because of the strict power limitations that exist in space, further improvements are required to make space-based laser manufacturing feasible.

EXPERIMENTAL SETUP AND PROCEDURES

Laser System and Other Experimental Equipment

Laser Equipment

The experimental setup for all laser-processing experiments is shown in Figures 6a and b. This includes the laser equipment, motion controller, focusing optics, shield gas delivery and powder delivery systems. The laser used in the present study was a 50 W Spectra-Physics Nd:YAG laser, which emits a 1064nm wavelength beam (specifications in Appendix E). From the specifications it can be seen that the laser can be operated both in pulsed and continuous wave mode. When used in the pulsed mode, the pulse repetition frequency (PRF) can be varied from 0.2 to 80 kHz. Additionally, the average laser power decreases slightly from 50 kHz to about 10 kHz and then shows a drastic decrease at much lower PRFs (e.g., <10 kHz). The pulse energy per shot increases with shorter pulse durations and decreasing PRFs. In continuous wave (CW) mode or operating at a high PRF, the laser is amenable for processes involving melting and cladding. Although the average power of the laser decreases at lower PRF's, the total energy per pulse increases. This makes the laser more suitable for other operations such as cutting and drilling. The empirically obtained laser power characteristics, measured as a function of diode current (from 20 to 46.3 A) for both CW and pulsed mode (50kHz of pulse repetition frequency), are shown in Figures 7a and b. It is also apparent from these findings that there is a threshold value of diode current (20 A) above which the measured laser power increases in a non-linear way for both CW and pulsed beam modes.

Optical System

The beam emitted from the laser is an unfocused beam of diameter \sim 2mm. The beam is then diverted downwards by a CVI Optics 2" diameter mirror, positioned at an angle of 45 degrees relative to the level of the laser (Fig 8). The mirror is specially constructed to reflect in excess of 99% of incident radiation at 1064 nm. The beam then passes through a 1" diameter plano-convex lens of focal length, f=100mm. The now-focused beam then impinges

on the test coupon, which has been placed at some optimum focal position of about 100 mm where the laser beam energy density was found to be the maximum. The laser beam was always viewed after wearing safety goggles (with optical density of 4.5 and above) and using an infrared viewer. The infrared viewer was a hand-held type of model #7215 made by Electrophysics Corporation. Additionally, "Zap-It" papers were used for beam alignment. For beam alignment, the laser was operated in continuous wave mode with approximately 20 A of diode current with a minimum of power for safety reasons. It is also recommended that laser beam alignment should not be conducted in the pulsed mode with very high laser power densities, especially when attempting to focus the beam.

Positioning Equipment

The test coupon itself is mounted and clamped onto a Velmex two-axis, computer-controlled linear stage. This arrangement allows movement of the target without risking disruption of the optical focusing equipment. The positioning stage has a number of different features (Table 3), including the capability of movement in both the x and y directions in steps as small as 5-8 mils. In the z direction, the precision is 20-30 microns (Table 3). Preliminary calibration of the linear speed along the x-axis is shown in Figures 9a-c.

The motion of the stage is controlled by software provided by Ability Systems, Inc, operating from a Windows 98 equipped, Pentium-class PC. This software, called Indexer LPT software, employs Hewlett-Packard Graphics Language (HPGL) commands to control the stage in much the same way as an old-style plotter is controlled. The HPGL commands are transmitted via 2 extra parallel port cards installed in the PC. The corresponding cables are attached to a translator/controller unit, which in turn, controls the actual operation of the stage step motors. Through a command-line programming interface, the Indexer LPT software also allows the stage to move in complex 2 and 3 dimensional patterns as desired.

Shield Gas and Ventilation Systems

The test coupons were bathed in a laboratory grade helium (He) gas envelope, delivered at a rate of 10 SCFH. Exhaust gases and metal vapors were collected by a custom-designed ventilation system (Figure 10). The intake was placed approximately 2 feet above the target area and was designed to allow for the collection of He without diverting the flow of shield gas before it could bathe the target area. The waste vapors and gases were then collected by the building ventilation system.

Powder Delivery System

The powder delivery system used in this experiment was a gravity-fed, manually controlled copper tube, with a powder reservoir located at the top of the tube (Figure 11). The powder reservoir was a 15cm long, 3cm diameter, transparent acrylic tube. This feature allowed the operator to quickly determine the amount of powder remaining in the feeder. At the top of the acrylic tube, a custom-machined fitting made of nylatron plastic and equipped with a rubber o-ring, acted as a stopper. The stopper was drilled through its longitudinal axis and equipped with a pneumatic fitting on its top surface. A tube from the shield gas bottle supplied He to the powder mixture to reduce the possibility of any premature reaction of the powder. This arrangement also allowed the shield gas to be delivered directly to the site of the powder-laser beam interaction while minimizing the risk of powder displacement from the larger, adjacent shield gas flow tube.

The acrylic tube was mounted on a machined brass sleeve that was tapered internally to retard the flow of the powder. Soldered to the bottom of the brass sleeve was a ¼" O.D. copper tube, equipped with a manually controlled butterfly valve. The valve was placed in the flow path, midway between the brass sleeve and the tube exit. The placement of the valve allowed the flow of the powder to be adjusted without forcing the operator to come in close proximity of the laser beam. The copper tube was of sufficient length and flexibility to place the tube exit within 3mm of the target/ laser beam interaction point. Copper was chosen as the feed tube material because of its high thermal conductivity/ reflectivity. These properties would allow the tube to have a higher resistance to any effects caused by the unintended impingement of the laser beam on the tube.

Experimental Materials Preparation and Analysis

Test Specimens

Three different alloys, Ti-6Al-4V, Al-2219, and Al-2090 were received as sheet/plate from NASA Langley. Their thermophysical properties are shown in Appendix G. These as-received samples were first partially sectioned using a power hacksaw into small specimens of approx. ½" square. These specimens were then polished, etched and subjected to different analysis techniques in order to ascertain the properties of each in the condition in which it was received (Figures 12a-e and Figures 13a-c). The rolled texture was observed in Ti-6Al-4V and the age-hardened microstructures were apparent for Al alloys.

Substrate Test Coupons

The as-received bulk material was also sectioned into a number of 2" x 4 ½ " test coupons. The sectioning was accomplished with an industrial, water/oil cooled band saw, operating at an extremely low cutting speed of less than 50 blade-feet/min to prevent warping of the test coupon. Each specimen was then rinsed with tap water, and cleaned in an ultrasonic bath using acetone followed by isopropyl alcohol, each for a period of five minutes. The specimens were then allowed to air dry before being placed in plastic bags in order to minimize contamination.

Alloy Cladding Powder

There were two alloy powders used in this experiment. The first powder used was Ti-6Al-4V obtained from Starmet Corporation. This powder ranged in size from 45 to 300 µm. The other alloy powder (Al-2219) was mechanically blended from elemental powders obtained from the Alfa-AESAR Corporation (Table 4). Due to the pellet nature of zirconium, it was not added to the other alloying elements in the elemental powder form. Due to its low concentration, this is not especially critical, considering that especially in laser processing produces greater solidification rates with grain refinement.

Sectioning, Polishing & Etching

The laser-processed test coupons were sectioned perpendicular to the process track for microscopy and other analyses. Each sample was in turn polished successively by 180, 240, and 400-grit SiC sandpaper, using water as lubricant. The samples were then chemically etched by a solution of 10ml HF acid, 5ml HNO₃ acid and 85ml H₂O for 30 s to 1 minute. Each specimen was then rinsed with tap water, and cleaned in an ultrasonic bath using acetone and then isopropyl alcohol respectively, each for a period of five minutes. The specimens were then allowed to air dry before being placed in holding containers in order to minimize contamination.

Digital Imaging

The visual records of the as-processed samples were obtained using both Nikon and Kodak digital cameras (Figures 14a-1).

Optical Microscopy (OM)

The processed test coupons were examined and imaged with a Nikon optical microscope, using a variety of magnification settings (Figures 15a-m).

Scanning Electron Microscopy (SEM)

The microstructures of the test specimens and of the substrate test coupons were investigated using a Scanning Electron Microscope (SEM) set at an operating voltage of 15kV and various magnification settings ranging from 88x to 9000x (Figures 16a-p).

Secondary Ion Mass Spectroscopy (SIMS)

Two of the selected samples (laser nitrided Ti-6Al-4V and Al-2219 alloyed on to Ti-6Al-4V substrate) were subjected to analysis by use of the Secondary Ion Mass Spectrometry (SIMS) technique. In this instance, Physical Electronics manufactured the SIMS machine used and the operating voltages for both positive (for heavy elements) and negative SIMS (for detecting light elements such as nitrogen and oxygen) were set to 15kV. The results are demonstrated in Figures 17a-u.

- Energy Dispersive Spectroscopy (EDS)

The same samples used for SIMS were also subjected to analysis by the use of Energy Dispersive Spectroscopy technique. The results of this analysis are demonstrated in Figures 18a-y.

Microhardness and Toughness Testing

The microhardness characteristics of the same samples used for EDS and SIMS (Table 5) were obtained using a Shimadzu Type M Vickers Microhardness Tester. The test loads used were 300, 500 and 1000g for a static loading time of 15 seconds. The visual search for cracking under the static indentation loads was used as a qualitative measure of toughness for the tested samples.

X-Ray Diffraction (XRD)

The X-Ray Diffraction (XRD) analysis (Figures 19a-b) was performed on the same samples using both Phillips and Rigaku model x-ray diffractometers. In both cases, the anode was a standard Cu anode with a wavelength of 1.54056.

Å. The operating voltage and tube current were set to 40 kV and 30mA, respectively. The scan angle range was from 10 to 130 degrees with a step size of 0.05 degrees, and a step time of 1s.

RESULTS AND DISCUSSION

Experimental Parameters

Preliminary energy calculations of the three experimental alloys are given in Appendix G. A number of different experimental parameters were varied in order to obtain the research findings contained herein. Some examples of these parameters include diode current, laser power, pulse repetition frequency, and process speed. These parameters and the observed effects of laser/material interaction mechanisms such as melt quenching, cladding/alloying and cutting, and the plasma characteristics of each experimental run, are listed in Tables 6a-w.

Laser Processing of Ti-6Al-4V

The results of the laser melt-quenched Ti-6Al-4V samples with or without He assist gas showed the following trends: In the case of continuous wave laser processing experiments, it is evident from Figure 20a that as the laser power was increased by increasing the diode current, the widths of melt and the heat affected zones (as evidenced by the different color zones) were found to increase.

The results of laser processing experimental runs performed at various pulse repetition frequencies on Ti-alloy samples at fairly constant diode current are shown in Figure 20b. It was evident that at lower pulse repetition frequencies, the laser pulse energy is high enough to cause sufficient cutting of the samples (due to vaporization) as opposed to melting. The threshold PRF was between 1 (Figure 15a) and 5 KHz for the Ti-alloy samples in this study. The cutting mechanism also produced blue plasma accompanied by significant noise as opposed to white plasma. The color variations of the plasma also provided supporting evidence of the physical mechanism change from melting to cutting in the test samples. The approximate heights of the plasma in select cases including the color changes above the sample surface were listed in Tables 6a-w. The plasma color observations were made after viewing the laser/sample interaction regions with the help of safety goggles (as always!!). Some of the blue colored heat affected regions in laser melted Ti-6Al-4V indicated to a certain degree the insufficient shielding in the process and also undesirable plasma heating of the substrate. Therefore it is essential to consider the right shield gas flow conditions to minimize such a heating effect. It was also noted that when the laser pulse frequency was increased, the extent of heat-affected region was also increased.

The results of cladding experiments with pre-placed Ti-6Al-4V alloy powder were also very encouraging. The clad beads looked fairly continuous, well bonded, and crack-free. At constant laser power, the increase in process speed was found to decrease the processed depth.

The only surface cleaning treatment for the experimental Ti-plates was done using acetone and isopropyl alcohol in conjunction with an ultrasonic cleaner before the actual tests. The experimental results provided encouraging evidence that the laser beam absorption properties (hence energy coupling) of the Ti-alloy samples were adequate in the as-received condition. This eliminates the need for any other absorbent coatings to be placed on the substrate. This was verified both in laser melt-quenched and laser cladding experiments by pre-placing a thin layer of alloy powder on the substrate. Without He gas, the observed plasma size above the surface was found to increase slightly with some smoking streaks from the interaction zones. Figure 14d shows the heat-affected zone without He as compared to that with He. This smoking effect was minimized by using He shield gas, flowing at a rate of 10 SCFH at the source.

The XRD results (Figure. 13a) of the as-received Ti-6Al-4V microstructures (Figure. 12a) revealed the presence of α-Ti (HCP) and β-Ti (BCC) phases. Additionally, the relative intensity variations and absence of some peaks (e.g., β-Ti) indicated the texture effect. The preliminary observations on Ti-samples indicated columnar dendritic morphology in the laser melt-quenched microstructures (Figure. 16b) with spacings of ~2-6 microns. The microstructures of the pre-placed 2219 alloy on Ti-6Al-4V substrate also revealed grains of (Figures. 16c, d, e, and f) of 5 to 10 microns. This is indicative of the presence of higher cooling rates even at low process speeds, unlike traditional casting or heat treatment. The interface between the Al-2219 alloyed layer and the Ti-substrate was free from any noticeable cracking or porosity. This demonstrates that the He shielding was adequate for the pre-placed powder coating experiments. The hardness tests (Table 5) revealed a substantial increase in the hardness of Al-2219 alloyed coating on Ti-substrate by a factor of ~5 as compared to that of as-received Ti-6Al-4V. These results are potentially applicable for the production of in-situ wear and corrosion resistant high temperature coatings made of intermetallic phases such as aluminides, without any vacuum requirement. The SIMS results revealed the presence of Cu, Ti, and Al in the coating regions, which was confirmed by the EDS results (Figures. 180-y). The XRD results indicated the possible presence of compounds such as Cu₃Al and/or Al₂Ti, for example. For further confirmation, a more detailed microchemistry analysis of various phase constituents and XRD scans at much lower scan rates, will be carried out in Phase II of this study.

- The hardness results of Al-2219 alloyed coatings (Table 5) provided no evidence of cracking under the test loads used in this study. This indicates the possession of significant toughness in addition to being much harder.
- The laser nitrided samples shown in Figures. 141 and 16g-i demonstrate the formation of outer and inner layers (2 to 3 distinct regions) in the microstructure. Please refer to Table 6v for laser nitridation process conditions. Figure. 18a n show the EDS results of the different nitrided regions. Indicated are the presence of the major element Ti and a small amount of Al. The XRD results shown in Figure. indicate the presence of variable stoichiometry compounds of Ti and N (e.g., Ti₃N_{2.X}). Additionally, the laser nitridation process parameters need to be optimized to eliminate cracking. For example when the laser was used in the pulsed mode (10 kHz), cracking was observed where it was absent in cw mode. This can be explained due to minimization of the stresses incurred during the less severe quenching of the cw mode (Figure. 16j). Of course, one can also produce crack-free coatings by using appropriately higher PRF for example.
 - The SIMS results revealed a similar trend of Ti-enrichment in the outer regions, with little Al present (Figures. 17m –u). The designation "Area A" in these figures, refer to the outer layers of the nitrided zone. To detect elements such as nitrogen and oxygen (e.g. <Z=20), negative SIMS was performed and is shown. SIMS results were also obtained for Areas B, C and D respectfully, which refers to progressively deeper layers, and ending with the untreated substrate.

Laser Processing of Al-2219

One of the technical strategies used to increase the initial coupling of the laser energy with the substrate material, is outlined below. The surface of the as-received Al-2219 test coupons was sprayed with a layer of energy absorbing black paint before several experimental runs. This practice of using energy absorbent coatings is fairly common when, for example, a CO₂ laser is used for heat treatment of steel surfaces. This is due to the higher reflectivity exhibited by the untreated steel surface.

In the present study, the laser process conditions used were higher laser power (at a diode current of 46.3 amps) and a constant process speed of 0.5 ipm in both cw and pulsed (PRF of 1 kHz) wave modes. The results yielded interesting melting and cutting on the highly reflective Al-2219 alloy. The results were not nearly as interesting for Ti-6Al-4V. The laser was successful in melting the Al-2219 (Figures. 15e, g and Figure. 16k) using highly energetic

pulses without pre-placed powder. The microhardness (Table 5) of the laser melt-quenched layer was found to be slightly higher than that of the untreated material. The processed layer depth was very small as compared to those obtained with Ti-6Al-4V and Al-2090 due to the much higher conduction losses of Al-2219. Figures. 15c, 15f, 15h and Figures 16 l and m reveal the effect of low PRF runs in which cutting was found to occur. Figure. 15 d shows both the central groove and laser melt-quenched edge microstructures of black painted Al-2219. Preliminary cladding results of Al-2219 using cw mode and by pre-placing the alloy powder showed some bonding after processing (as shown in the visual record). With high power density cw lasers, it is possible to increase the processed layer depth using powder melting onto an Al-2219 substrate. The XRD results (Figure. 13c) of the asreceived Al 2219 microstructures (Figure. 12b and 12c) revealed the presence of α-Al (FCC) as the major phase present.

Laser Processing of Al-2090

Laser melt-quenched results of Al-2090 microstructures as observed from the top surface, are shown in Figures 15i – m as well as in Figures 16o-p. The rapidly solidified cellular grain boundaries are apparent, as is the lack of porosity and cracking, as evidenced from these micrographs. It was also found that the black painting of the surface prior to laser processing was not needed for Al-2090 due to its relatively lower thermal conductivity (as compared to Al-2219). The XRD results (Figure. 13b) of the as-received Al-2090 microstructures (Figure. 12d and 12e) revealed the α-Al (FCC) as the major phase.

Summary of Phase I Research

In Phase I of the research project, what have been presented are the feasibility findings, which will be used for further exploration in Phase II. The laser given by the vendor to the PI's project is a brand new state-of-the-art laser, which has not yet been used in conjunction with manufacturing. This laser has far superior characteristics, which are suitable for a melting/cladding study as compared to any of the other available lasers at ARC-ODU at the time of experimentation. That is one of the major reasons to choose this type of laser for this challenging and exploratory study. It is also noted that all of the above-mentioned experiments were performed using a 100 mm focal length lens, a much smaller multi-mode (e.g., with M² of 14 or so) laser beam size (e.g., 30 to 100 micron size, and a larger depth of field (constant power density within this depth of field). The process speeds used in this study are relatively slow due to the lower power densities possible with this laser. The reason why we chose a larger focal length is to accommodate the shield gas and powder delivery nozzles. This arrangement is also advantageous in that it is possible to explore other focal length lenses as well. For example, the focused laser power densities and energies can be increased using a shorter focal length lens (e.g., 50 mm).

Phase II

Phase II is planned to be a continuation of the work begun in Phase I. Specifically, further laser cladding experiments and free-form fabrication of samples for Ti-6Al-4V and Al-2219 have been proposed. A more efficient powder delivery method, and a continuous wave Nd:YAG laser with higher average power (e.g., 500W) and relatively larger beam diameter, are planned to be utilized. It is expected that this will help to increase the processed layer depth and width by substantially increasing the melting of the cladding alloy powder. These changes also have the added benefit of increasing the process speed by a substantial margin. Typical cost comparison of selected available laser equipment is provided in Appendix H for easy reference. The initial experiments in Phase II are planned to be performed using a vendor-supplied laser, loaned to the PI. This is viewed as the most cost-effective method of experimentation before a recommendation is issued (if needed) relating to the purchase of the primary laser to be used in future R&D. The PI is also currently seeking cost-sharing assistance from the ODU College of

Engineering in this regard. Additional details are to be provided in a separate Phase II proposal submitted by the PI to NASA Langley Research Center.

CONCLUSIONS FROM THE RESEARCH FINDINGS

The following conclusions were drawn from the experimental results of the present research investigation:

- Ti-6Al-4V samples can be successfully laser melted/cladded using the pulsed/cw Nd:YAG laser
- Dense, crack-free and well-bonded melt-quenched layers/coatings were produced in single layers
- Key laser variables such as laser power, pulse repetition frequency and process speed had direct influence on the laser-processed geometry such as melt width, kerf width and heat affected zone.
- The low repetition frequencies of the pulsed laser (e.g., 1 kHz) produced cutting in both Ti-6Al-4V, Al-2219, and Al-2090, whereas melt-quenching was obtained using high PRF (20 kHz or higher) and continuous wave beam modes
- Ti-6Al-4V and Al-2090 samples were laser processed without any energy absorbent black coatings
- Absorbent coatings were found to be beneficial to increase the initial laser energy coupling of Al-2219 samples
- The laser processed microstructures were much finer than those of the untreated substrate materials, indicating higher cooling rates
- In-situ production of hard aluminide and nitride coatings, with excellent hardness properties, on Ti-6Al-4V was demonstrated

PROJECT ACCOMPLISHMENTS

The following list provides some of the project accomplishments under this research grant:

New Research Lab Establishment
Fabrication of Key Experimental Components
Acquisition and Testing of Essential Motion Controller
Groundwork for the Preliminary Research Findings and Phase II Study
Collaborative ARC/ODU Paper for STAIF-2001 Conference
Educational and Research Support from NASA Langley

SUMMARY AND FUTURE OF LASER MANUFACTURING IN SPACE

With launch costs and power constraints in space in mind, the development of a multifunctional tool would have potential for actual application in short- or long-term space missions. Laser power efficiencies have been improved from 2% to as much as 40%. Fiber optic delivery has greatly increased the flexibility of laser manufacturing by expanding the ability to treat complex geometries and inaccessible areas with less difficulty. Shorter wavelengths, higher reliability, lower input power requirements with greater efficiency, low maintenance requirements, the ability to use fiber optic beam delivery, and the compact size of diode pumped solid-state lasers, are all factors which will potentially serve the needs of in-space manufacturing. They are also suitable from a multifunctional perspective, but significant improvements in power efficiency are necessary for applicability in space missions. These lasers are inherently well suited to be used as a multifunctional tool because they have characteristic wavelengths and power outputs suitable for most laser materials processing techniques. The continuing development of these types of lasers is currently addressing manufacturing issues, which will also impact the ability to modify these systems for space-based manufacturing. Establishing an in-space laser manufacturing capability will require significant improvements over state-of-the-art laser technology in power efficiency, maximized life with minimal maintenance, and the ability to operate using DC input power.

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Table 1. Typical Processing Parameters and Results of Laser-Processed Al, Ti and Cu-Based Materials.

- 1

Material	Type of Laser	Laser Type Used	Typical Process Parameters	Comments	Reference
7175 Al substrate/ Cr alloying	Laser surface alloying	Continuous wave CO ₂	Power Density: 120-260 kW/cm ² Interaction Time: .043 s Scanning Rate: 5, 10 mm/s Remelting Rate: 10, 20 mm/s	VHN=155 (Cr-alloyed Al) VHN = 300 (Cr-alloyed 7175); reduced porosity after remelting; improved pitting corrosion resistance with Cr; Al, Cr, Al, Cr, & α-Al phases	Almeida, et al., 1995
Ti substrate/ Al alloying	Laser surface alloying	Continuous wave CO ₂	Power: 1.8 kW Beam Diameter: 3 mm Process Speed: 3-20 mm/s Powder Feed: 0.03-0.11 g/s	Alloy is homogeneous & reproducible; alloy build-up thickness: 0.2 -0.9 mm; 17-80% Al concentration with increase in powder feed rates	Abboud, et al., 1991
Ti-6Al-4V substrate/ C alloying	Laser surface alloying	Continuous wave CO ₂	Laser Power: 3 kW Process Speed: 40-100 ipm	Dendritic & cellular structures; roughness decreases from 50 to 10 microns & chemical homogeneity improves as # of scans incr. from 0 to 4	Bharti, et al., 1989
6061 Al substrate/ Ni, Cr alloying	Laser surface alloying	Continuous wave Nd:YAG	Power Density: 30 J/mm ² Beam Diameter: 2 mm Scanning Rate: 6 mm/s	Fretting wear & coefficient of friction reduced significantly	Fu, et al., 1998
Pure Al substrate/ Ni alloying	Laser surface alloying	Pulsed Nd:YAG	Power: 120 W Pulse Width: 4 ms Repetition Rate: 10 Hz Speed: 2.1 mm/s % Overlap of Tracks: 65%	α-Al, lamellar eutectic of α-Al & Al ₃ Ni and primary Al ₃ Ni dendrites depending on the Ni contents (6-15% in alloyed zone)	Das, et al., 1994
Cu substrate/ Al alloying	Laser surface alloying	Pulsed Nd:YAG	Power: 160W Beam Size: 0.45 mm Pulse Duration: 2 ms Frequency: 10 Hz Process Speed: 1 mm/s	Non-ferrous martensitic microstructures; improved mechanical strength & wear resistance	Nagarathnam, 1995- 1997

Table 1 (Contd.). Typical Processing Parameters and Results of Laser-Processed Al, Ti and Cu-Based Materials.

Reference	Nagarathnam, et al., 1996	Kelly, et al., 1995, 1998	Katsuna, et al., 1993	Coyle, et al., 1987	Ramaswamy, et al., 2000
Comments	Homogeneous single phase α-Fe-Cr-Al microstructure with sufficient yttrium; 15-150 microns grain sizes; improved high temperature oxidation resistance at 1200°C	Fine grained microstructure of primary α-phase and eutectic interdendritic phases; increased hardness, toughness & improved sliding wear resistance	Penetration depths: 3.25, 4.30, 6.5, & 7.10 mm	Successful welding	Reduced weld hardness due to solutionizing effect; cellular dendrites in fusion zone with equiaxed grains at weld center; successful autogenous butt welding; no post weld heat treatment
Typical Process Parameters	Laser Power: 0.6-2.4 kW Process Speed: 0.53-4.23 mm/s Powder Feed Rate: 0.083 g/s Beam Delivery: fiber optic	Power: 2.3-2.5 W Process Speed: 14-16 mm/s Powder Feed Rate: 0.7 g/s Specific Energy: 34-48 J/mm ²	Power: 3, 4, 6 and 8 kW	Power: 20 W Pulse Duration: 7 ms Frequency: 1 Hz Shield Gas: Ar	Power: 2 kW Weld Speed: 42 – 63 mm/s Beam Diameter: 0.5 mm Sheet metal thickness: 1 mm Beam Delivery: 1 mm fiber optic
Laser Type Used	Continuous wave Nd:YAG	Continuous wave CO ₂	Continuous wave CO ₂	Pulsed Nd:YAG	Continuous wave Nd:YAG
Type of Laser Process	Laser surface cladding	Laser surface cladding	Laser beam welding	Laser beam welding	Laser beam welding
Material	316L stainless steel substrate/ MCrAlY alloy	Cast 390 & 333 Al- Si alloy substrate/ Sn & Mn- Bronze alloy coatings	2219, 5083, 6063 Al	6061, 4047 AI	6111-T4 Al

Table 1 (Contd.). Typical Processing Parameters and Results of Laser-Processed Al, Ti and Cu-Based Materials.

Material	Type of Laser Process	Laser Type Used	Typical Process Parameters	Comments	Reference
8090 AI	Laser beam welding	Continuous wave CO ₂	Power: 1.5-3.8 kW Scanning Rate: 20-120 mm/s Beam Diameter: 0.25 mm Coaxial He Flow: 40 l/min Sheet Thickness: 2-3.2 mm (60 mm/s-full weld penetration) 4 mm (60 mm/s-partial weld penetration)	2-5 micron secondary dendrite arm spacing; columnar & equiaxed grains; fine scale δ' (Al ₃ Li) ppts. within α-phase; increased hardness; porosity along weld centerline	Whitaker, et al., 1993
Al-30% SiC particulate composite	Laser micro- machining	Diode- pumped Nd:YAG	Wavelengths: 532 nm (green) at 35 W 1064 nm (IR) at 85-90 W Power: 35-90 W Modes: Q-switch+ Mode Lock: 150 ps pulses (13 ns apart) at 76 MHz Q-Switch (5 kHz & 300 ns) Process time: Up to 180 s Assist Gas: 45 psig O ₂	Straight & angular holes; no recast layer & heat affected zone under optimum conditions; hole depths: 85-420 microns; hole width: 30-425 microns; higher material removal rate at 532 nm wavelength	Nagarathnam, et al., 1997

Table 2. CO₂ Laser Materials Processing Parameters for Various Surface Engineering Applications.

Surface Engineering Process Laser Surface Transformation	Density (Power/Beam Area), kW/cm²	Specific Energy (Power/Beam Size Speed), J/cm²	Laser Beam Interaction Time (Beam Size/Speed), s
Hardening	0.0-40	00-00	7-10-0
Laser Surface Melt-Quenching	2.5-7000	0.3-200	0.001-9
aser Surface Alloying	3-40000	1-200	0.001-10
aser Surface Cladding	3-150	1-100	0.1-10
aser Melt-Particle Injection	30-310	5-30	0.04-0.55

Table 3: Specifications of Velmex Bi-Slide™ 2-Axis Linear Stage

Table 4: Composition of Mechanically Blended Al-2219

Wt of Component Elements (g)	93.08	6.304	not used	not used	not used	not used	0.0605	0.1157	not used	0.1835	300 T
% 7M	93	5.8 - 6.8	Max 0.3	Max 0.02	0.2 - 0.4	Max 0.2	0.02 - 0.1	0.05 - 0.15	Max 0.1	0.1 - 0.25	
Component	Al	Cu	Fe	Mg	Mn	Si	T	Λ	Zn	Zr	Total:

Table 5: Microhardness Test Results

	Load	Time	Trial	Horizontal		Ha	ardness Valu (kg _f /mm²)	es
Material	(g)	(s)	Number	Distance (µm)	Distance (µm)	Obtained	Calculated (Average)	Expected
As-Received Ti-6Al-4V	500	15	1	49	48.6	389	378	396
	500	15	2	48.5	52.5	363		
	500	15	3	51	47.7	381		
	1000	15	1	68.3	70.5	385	378	396
	1000	15	2	71.4	69.8	372		
As-Received Al-2090	300	15	1	60.8	59.7	153	151	
d i	300	15	2	66.1	58	145		
	300	15	3	62	58	155		
As received Al-2219	300	15	1	62	57.5	156	154	149
	300	15	2	61	60.4	151		
	500	15	1	79.3	80.3	146	147	149
	500	15	2	78.3	76.6	149		
	500	15	3	80.1	79.3	146		
Laser Nitrided Ti-6Al-4V - Substrate (11-01-2000 TC4 R1)	500	15	1	45.4	47.2	433	452	7.00 minutes (1.00 minutes) (1.00 mi
	500	15	2	45.9	44.5	454		
	500	15	3	43.5	45.4	469		
Laser Nitrided Ti-6Al-4V Inner Layer of Nitrided Zone (11-01-2000 TC4 R1)	1000	15	1	34.1	33.2	1639	1582	
	1000	15	2	36.0	33.7	1527		
	1000	15	3	34.4	34.2	1580		
Laser Nitrided Ti-6Al-4V Outer Layer of Nitrided Zone (11-01-2000 TC4 R1)	1000	15	1	28.1	30.2	2180	2180	
Al-2219 PPP on Ti-6Al-4V Substrate – Alloyed Zone (10-30-2000 TC4 R3)	300	15	1	18.3	19.6	1547	1768	
	300	15	2	19.0	19.1	1529		
	300	15	3	15.9	15.7	2228		
Al-2219 PPP on Ti-6Al-4V Substrate Substrate Region (10-30-2000 TC4 R3)	300	15	1	39.7	38.5	364	370	
	300	15	2	37.8	40.0	368		
	300	15	3	37.0	39.5	379		
Laser Melted Al-2219 - Kerf	500	15	1			493	494	
	500	15	2			487		
	500	15	3			501		

Tables 6a-w: Experimental Parameters and Observed Effects

Table 6a

Date: 10-03-2000 Test Coupon 1	ted Ti-6Al-4V Process Type: Surface Melting	Shield Gas: None Process Speed (v): 100 mm/min	Comments	Zigzag pattern observed	Melting observed	Melting observed	Melting observed	1mm yellow plasma observed	
Date: 10-03	reated Ti-6A		HAZ Width (µm)		1193.8	1701.8	1727.2	1930.4	
	Material: Untreat	Beam Mode: CW	Diode Current Ave. Laser Power HAZ (W) (W)	9.6	22.4	32.68	38.2	41.4	
			Diode Current (A)	25	30	35	40	46.3	
			Run Number	1^	2^	3^	4	5.	

^ Measurements obtained from photographic analysis

Table 6b

Date: 10-03-2000 Test Coupon 2	Material: Untreated Ti-6Al-4V Process Type: Surface Melting	Shield Gas: He @ 10 SCFH Process Speed (v): 100 mm/min	Comments	Melting observed					
Date: 10-(treated Ti-6		HAZ Width (µm)		482.6	787.4	1371.6	1447.8	1549.4
	Material: Un	Beam Mode: PRF=20 kHz	Diode Current Ave Laser Power HA (A)						~35
		Beam M	Diode Current (A)	20	25	30	35	40	46.3
			Run Number	1>	2^	3^	4	5^	v9

^ Measurements obtained from photographic analysis

Table 6c

Date: 10-03-2000 Test Coupon 3	Material: Untreated Ti-6Al-4V Process Type: Surface Melting	Beam Mode: PRF=50 kHz Shield Gas: None Process Speed (v): 100 mm/min	Comments	No effect	Some plasma observed	Some plasma observed			
Date: 10-03	reated Ti-6A	kHz Shiel	HAZ Width (µm)		533.4	787.4	1219.2	2032.0	
	Material: Uni	am Mode: PRF=50	Diode Current Ave. Laser Power HA (W)	0.2	6.8	20.5	30.9	34.1	
		Be	Diode Current (A)	20	25	30	35	40	
			Runnumber	1^	2^	3^	4	5^	*

^ Measurements obtained from photographic analysis

Table 6d

			Date: 10-03-2000		Test Coupon 4	14		
		Material: Un	Material: Untreated Ti-6Al-4V		cess Type	Process Type: Surface Melting	Melting	
		Shield Gas: He	He @ 10 SCI	@ 10 SCFH Process Speed (v): 100 mm/min	ss Speed	(v): 100 m	m/min	
Run Number	The state of the s	Diode Current Ave Laser Power (A) (W)	PRF (kHz)	Melt Width (µm)	HAZ Depth (µm)	Kerf Width (µm)	Kerf Depth (µm)	Comments
+	46.3	~35	1			70	112.4	Sparks observed 10mm plasma observed
2+	46.3	~35	S			331.2	250.4	10mm plasma observed
3+	46.3	~35	10	342	35.6			
++	46.3	~35	20	380	42.8			No sparks observed <1mm plasma observed
\$+	46.3	~35	40	432	87.2			<1mm plasma observed
+9	46.3	~35	09	356	62			<1mm plasma observed
7+	46.3	~35	80	233.6	42			Erratic track pattern

+ Measurements obtained from optical microscopy

t Table 6e

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			<u></u>						
	ice Melting	Focal Position: 114.3 mm	Comments						Wider melt zones at lower speeds Warping of substrate Oxidation observed No sparking
n(s): 1 & 2	Type: Surfa	al Position	HAZ Width (μm)			3362	3838	5427	
10-09-2000 Test Coupon(s): 1 & 2	4V Process 7	Shield Gas: None Foo	Travel Distance (mm)	8.05	38.1	25.4	25.4	25.4	
Date: 10-09-2000	Material: Untreated Ti-6Al-4V Process Type: Surface Melting	l	Process Speed (mm/min)	8.05	25.4	50.8	25.4	12.7	nts
Da	Material: Unt	Beam Mode: CW	Diode Current Ave Laser Power Process Speed (A) (mm/min)	41.4	41.4	41.4	41.4	41.4	General Comments
			Diode Current (A)	46.3	46.3	46.3	46.3	46.3	
			Run Number	1^	2^	3^	4	\$>	

^ Measurements made by photographic analysis

Table 6f

Date: 10-09-2000 Test Coupon(s): 3	Material: Untreated Al-2219 Process Type: Surface Melting	Shield Gas: None Focal Position: 114.3 mm	(mm)	No observable HAZ	Run aborted	4 sellow plasma observed	
000 Test	19 Proc	as: None	Travel Distance (m	50.8	50.8	50.8	50.8
Date: 10-09-2(Intreated Al-22		Process Speed Travel (mm/min) Distance (mm)	8.05	25.4	25.4	12.7
	Material: U	Beam Mode: CW	Diode Current Ave. Laser Power Process Speed (A) (mm/min)	41.4	41.4	41.4	41.4
			Diode Current (A)	46.3	46.3	46.3	46.3
			Run Number	9	7	8	6

Table 6g

Coupon(s): 1	5Al-4V PPP Process Type: PPP Melting	CFH Focal Position: 114.3 mm	oltage Process Speed Trave Distance (V) (mm/min) (mm)	50.8 Melting observed	38.1 Overlapped into first track	25.4 1mm plasma observed	25.4 2mm plasma observed	25.4 Yellowish layer present on surface	Comments	Partial Width Measurement	Result due to discontinuous clad layer	
Date: 10-10-2000 Test Coupon(s): 1	strate and Ti-6	Shield Gas: He @ 10 SCFH	Process Speed (mm/min)	50.8	25.4	25.4	25.4	25.4	Clad Depth (µm)	12.7		
Date: 10-	VI-4V Sub	Shield G	Voltage (V)	10.73	10.73	10.73	10.73	10.73	Clad			
	Material(s): Untreated Ti-6Al-4V Substrate and Ti-6Al-4V PPP	Beam Mode: CW	Diode Current Ave Laser Power W (A)	41.4	41.4	41.4	41.4	41.4	Clad Height (µm)	70.0	8.4	68.8
	Material(1	Diode Current (A)	46.3	46.3	46.3	46.3	46.3	Run Number Clad Width (Lm)	264.7	40.5	449.6
			Run Number	1	2	3	4	w	Rum Number	3*	4 *	*\$

^{*} Measurements obtained from scanning electron microscopy

Table 6h

	Material(s): Untreated Ti-6Al-4V Substrate and Ti-6Al-4V PPP Process Type: PPP Melting	.3 mm	HAZ Width (µm)	Some melting observed Track is extremely narrow	5425 <1mm plasma observed	4953 <1.5mm white plasma observed	3325 <1.5mm white plasma observed
	Process	n: 114.	saniaus :		52	49	33
Coupon(s): 2	Al-4V PPP	V Shield Gas: None Focal Position: 114.3 mm	Process Speed Travel Distance (mm/min)	50.8	50.8	50.8	50.8
Date: 10-10-2000 Test Coupon(s): 2	strate and Ti-6	ld Gas: None	Process Speed (mm/min)	12.7	12.7	25.4	50.8
Date: 10-1	Ti-6Al-4V Sub	ode: CW Shie	Voltage 1	10.73	10.73	10.73	10.73
	ial(s): Untreated	Beam Mode: CW	Ave. Laser Power (W)	41.4	41.4	41.4	41.4
	Materi		Diode. Current (A)	46.3	46.3	46.3	46.3
			Run Number	9	7^	*	\ 6

[^] Measurements made by photographic analysis

			Date: 10-10-2000		Test Coupon(s): 3			
	Mat	Material(s): Untreated Al-221	ted Al-2219 Sul	19 Substrate and Al-2219 PPP		Process Type: PPP Melting	PPP Melti	gu
			Shield Gas: None		Focal Position: 114.3 mm	æ		
Run Number	Diode Current (A)	Ave. Laser Power (W)	Voltage (V)	Beam Mode (kHz)	Process Speed (mm/min)	Travel Processed Distance Width (mm) (µm)	Processed Width (µm)	Comments
10	46.3	41.4	10.73	CW	12.7	50.8		<.5mm white plasma observed No sparking present Powder layer was relatively thick
11^	46.3	41.4	10.73	CW	12.7	50.8	647	Powder layer thickness reduced on this run <.5mm white plasma observed
12^	46.3	41.4	10.73	CW	12.7	50.8	747	Performed along same track as 11 to see if remelting occurs
13^	46.3	~35	10.73	1	12.7	50.8	993	Thick powder layer <.5mm of blue plasma observed No sparking observed
14^	46.3	~35	10.73	10	12.7	50.8	827	<.25mm plasma No pushing of powder layer to side of track

1 Table 6i

^ Measurements made by photographic analysis

ا نے	_		-				_
		ting		Comments	Melting observed White plasma observed	Switched from shield gas to no shield gas in middle of run	Demo sample for NASA Sponsor
_		face Mel	m	HAZ Width (µm)	2311	2616	
. 1	ate: 10-13-2000 Test Coupon(s): 1	Material(s): Untreated Ti-6Al-4V Process Type: Surface Melting	Shield Gas: None Focal Position: 114.3 mm	Travel Distance (mm)	38	44.5	
 Table 6j	13-2000 Tes	Fi-6Al-4V P	Vone Focal	Process Speed (mm/min)	12.7		
-	Date: 10-	l(s): Untreated	Shield Gas:	Beam Mode (kHz)	CW		General Comments
- -		Material		Ave. Laser Power (W)	41.4		Gene
				Diode Current (A)	20		
-				Run Number		7	

^ Measurements made by photographic analysis

Table 6k

Date: 10-27-2000 Test Coupon(s): 1	erial(s): Black Painted Al-2090 Process Type: Black Paint Melting	Shield Gas: He @ 10 SCFH Focal Position: 114.3 mm	Voltag (V)	4 10.73 CW 12.7 25.4	.4 10.73 CW 12.7 25.4	.4 10.73 CW 12.7 25.4 High speed Test	35 10.73 20 12.7 25.4 Plasma Observed	 10.73 1 12.7 25.4 Cutting observed High pitched, audible noise present 	 -1mm blue plasma observed Cutting observed 10.73 2 12.7 25.4 High pitched, audible noise present Noise is higher pitched than previous run 	35 10.73 10 12.7 25.4 Little noise	Two passes on same track-melting 10.73 10 12.7 25.4 Two passes on same track-melting observed	poth Kerf Depth Kerf Width Track Width Comments (μm) (μm) (μm)	63.8 µm grain size (top central zone) 405.7 33.0 µm(near interface) 2-4 µm at interface	149.2 104.4	+	149.2 89.2 Measurement to be redone
			Voltage Beam Mode (V) (KHz)		3	8	3	10.73				Depth Kerf Width um			49.2 89.2	
	Material(s): Black	Shield	Ave Laser Power (W)	41.4	41.4	41.4	~35	~35	~35	~35	~35	Melt Width Melt Depth Ker (µm) (µm) (1	
			Diode Current (A)	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	Catalon Blockers				
			Run Number		2	3	4	v	9	A	В	Rum Number	4	\$+	+9	

⁺ Measurements obtained from optical microscopy * Measurements obtained from scanning electron microscopy

Table 6m

+ Measurements obtained from optical microscopy

			oe: Welding		Comments	<1mm plasma observed Audible sound present	<1mm plasma observed Audible sound present Lower pitched than previous run	Al foil was placed dull side up to enhance coupling
)	3): 1	Process Ty	n: 114.3 mm	Travel Distance (mm)	25.4	25.4	il was placed
	e 6n	Date: 10-30-2000 Test Coupon(s): 1	Material(s): Black Painted Ti-6Al-4V strip with Al foil Process Type: Welding	Shield Gas: He @ 10 SCFH Focal Position: 114.3 mm	Voltage Beam Mode Process Speed (V) (kHz) (mm/min)	12.7	12.7	Al fo
	Table 6n	10-30-2000	Ti-6Al-4V str	e @ 10 SCFH	Beam Mode (kHz)	10	20	
		Date:	k Painted	d Gas: Ho	Voltage (V)	10.73	10.73	4.00
			aterial(s): Blac	Shiel	Ave. Laser Power (W)	~35	~35	General Comments
1. The Health Leader Managacount the Hall Hall Hall			M _E		Diode Current Ave. Laser (A) Power (W)	46.3	46.3	Genera
	_				Run Number	-	2	

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			Date	: 10-30-2000	Date: 10-30-2000 Test Coupon(s): 2	3): 2	
PARTITION AND THE PARTITION AN		Materi	al(s): Bla	ck Painted Al-	Material(s): Black Painted Al-2219 Process Type: Cutting	Type: Cuttin	3 1
		Shiel	Shield Gas: H	le @ 10 SCFH	He @ 10 SCFH Focal Position: 114.3 mm	n: 114.3 mm	
Run Number	Diode Current Ave. Laser Voltage (A) Power (W) (V)	Ave. Laser Power (W)		Beam Mode (kHz)	Beam Mode Process Speed (kHz) (mm/min)	Travel Distance (mm)	Comments
_	46.3	~35	10.73	10	12.7	25.4	<1mm of blue plasma present Audible noise present
2	46.3	~35	10.73	.2	12.7	25.4	<1mm of blue plasma present Audible noise present
3	46.3	~35	10.73	15	12.7	25.4	<1 mm of white plasma present Quieter than previous runs
4	46.3	~35	10.73	20	12.7	25.4	<1 mm of white plasma present Quieter than previous runs
ĸ	46.3	41.4	10.73	CW	12.7	25.4	<1 mm of white plasma present Quieter than previous runs Less melting than previous runs

	tting		Comments	<1mm blue plasma observed Some audible noise present	<1mm blue plasma observed Some audible noise present	<1mm blue plasma observed No audible noise present	Intense blue plasma observed <1mm in height	Imm blue plasma observed No audible noise present	No effect	<1mm plasma observed No audible noise present f changed to 100 mm	More audible noise than #7 Better coupling f changed to 108mm	Cutting observed f changed to 109.5mm	<1mm plasma observed f changed to 114.3mm	
~	elting/Cu	114.3 mm	Travel Distance (mm)	25.4	25.4	25.4	25.4	25.4	25.4	25.4	25.4	25.4	25.4	
Toct Counon(c): 3	Process Type: Melting/Cutting	Focal Position: 114.3 mm	Process Speed (mm/min)	12.7	12.7	12.7	12.7	12.7	25.4	25.4	25.4	25.4	25.4	
T 000 00 10 10 T]	-	Beam Mode (kHz)	10	7.	20	1	15	CW	10	10	10	10	y
Dotor	Material(s): Untreated Al-2219	Shield Gas: He @ 10 SCFH	Voltage (V)	10.73	10.73	10.73	10.73	10.73	10.73	10.73	10.73	10.73	10.73	+ Measurements obtained from optical microscopy
			Ave. Laser Power (W)	~35	~35	~35	~35	~35	41.4	41.4	41.4	41.4	41.4	tained from
			Diode Current (A)	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	rements ob
			Run Number	1	7	3	4	S	9	7	∞	6	10	+ Measu

Table 6p

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^{*} Measurements obtained from scanning electron microscopy

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						ops	m to					
_						Recast layer and burr formation observed	Cellular grains observed from top	Z			Ð	
						mat	/ed	HA			rve	
				Comments		for	ser	No observable HAZ			Grooving observed	liec
						urr	qo	rva			o Su	r the
				ပြီ		q p	ains	pse			OVİ	1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
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_		ng/	3.	418.04 i.e.		2						sts le o ns
		[elti	114								L	f te
-	Date: 10-30-2000 Test Coupon(s): 3	reated Al-2219 Process Type: Melting/Cutting	Focal Position: 114.3 mm	l a								This series of tests was conducted on the reverse side of Test Coupon 2 Focal position in runs 7-10 was varied for effect
-)uo	y Pe	siti	Melt Width (µm)						ĸ	ن.	seri rev sitic
	oup!	SS T	Po	elf Wid (µm)						229.5	100.3	ii e
= }	C	S	oca	Me						``		T E
ع و	Tes	Ā	E									
Toble for (Confd)		6	Ħ									
		221	Š	pt .								
- E	Ş	 	10.	Melt Depth (μm)	125.5	44.6					167.0	
_	10-	ted	8		1.	4					Ī	
	te:	rea	He	<u> </u>								
-	a	Material(s): Unt	Shield Gas: He @ 10 SCFH				<u> </u>	(= ^				
		3	9 p									
_		jal(hiel	t)								
		ater	S	Kerf Depth (µm)	141.2	114.4		138.8			150.3	
		Σ̈́		달크	14	11		13			15	
-				2								S
												General Comments
_				8								, di
				Kerf Width (µm)	∞.	 ∞		œ.			6	Ŭ
-					126.8	50.8		136.8			53.3	era
	:			Ž								l e
				734-41-54	_	\vdash	-		-			
-				υ p e								
					+	5 +	3+	+	*9	*	10*	
_				Run Number		``			-			
				R								

Diode Current Ave. Laser
(KEZ)
10.73 CW
10.73 CW
10.73 50
10.73 30
10.73 20

	ding		h Comments			
	e: PPP Clad		Clad Depth (µm)		244.7	3
	Process Typ	14.3 mm	Clad Height (µm)		323.2	504.2
ate: 10-30-2000 Test Coupon(s): 4	Material(s): Black Painted Ti-6Al-4V Substrate & Al-2219 PPP Process Type: PPP Cladding	Shield Gas: He @ 10 SCFH Focal Position: 114.3 mm	Clad Width (µm)	546.8	712.1	737.9
ate: 10-30-2000	6Al-4V Substrate	: He @ 10 SCFH	Boundary Layer Thickness (μm)	39.9		
Q	inted Ti-	hield Gas	HAZ Width (um)	467.5		
	s): Black Pa	S	Melt HAZ Depth Depth (μm)	448.4		
	aterial(Melt Depth (µm)	288.3		
	M		Melt Width D ((µm)	307.44		
			Run	*	3*	5*

* Measurements obtained from scanning electron microscopy

			O D	ate: 11-01-20	Date: 11-01-2000 Test Coupon(s): 1	on(s): 1	
	M	aterial(s): Unt	reated Al-2	090 Substrate	Material(s): Untreated Al-2090 Substrate & Ti-6Al-4V PPP		Process Type: PPP Cladding
			Shield Ga	Shield Gas: He @ 10 SCFH	ŀ	Focal Position: 114.3 mm	
Run	Diode Current Ave. Laser	Ave. Laser Power (W)	Voltage (V)	Beam Mode	Process Speed (mm/min)	Beam Mode Process Speed Travel Distance (kHz) (mm/min)	Comments
	46.3	~35	10.73	10		25.4	
7	46.3	~35	10.73	15	12.7	25.4	>1mm of plasma observed Audible noise present Large amount of sparking present
ဗ	46.3	~35	10.73	20	12.7	25.4	>1mm of plasma observed Audible noise present Large amount of sparking present
4	46.3	~35	10.73	50	12.7	25.4	>1mm of plasma observed Audible noise present Large amount of sparking present
w	46.3	41.4	10.73	CW	12.7	25.4	>1mm of plasma observed Audible noise present Large amount of sparking present
9	46.3	41.4	10.73	CW	12.7	25.4	>1mm of plasma observed Audible noise present Large amount of sparking present

⁺ Measurements obtained from optical microscopy * Measurements obtained from scanning electron microscopy

Table 6r (Contd.) Date: 11-01-2000 Test Coupon(s): 1 Material(s): Untreated Al-2090 Substrate & Ti-6Al-4V PPP Process Type: PPP Cladding	Chield Cas: He @ 10 SCEH - Rocal Position: 114 3 mm
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Material(s): Untreated Al-2090 Substrate & Ti-6Al-4V PPP Process Type: PPP Cladding	Shield Gas: He @ 10 SCFH Focal Position: 114.3 mm	Omments						Shield gas turned off and turned on halfway thru runs 3 and 4 Smoke and sparks present during absence of shield gas Level of audible noise decreased as PRF increased
trate & Ti	10 SCFH	HAZ Depth (μm)	167.6	8.09	128.4			Shield
sqnS 060	ıs: He @	HAZ Width (µm)	302.4	170.8	241.2			
ntreated Al-2	Shield Ga	Clad Height (µm)				54.3	42.0	
Material(s): U		Clad Width Clad Depth (μm)				88.0	52.7	General Comments
Z		Clad Width (µm)				304.9	358.5	General
		Run	1+	2+	3+	*0	*9	

	Process Tyne: PPP Cladding	0	Comments	10mm of plasma observed Audible noise present Large quantity of small sparks present Powder was pushed to one side Track width 10mm NO SHIELD GAS	10mm of plasma observed Audible noise present Large quantity of small sparks present Powder was pushed to one side Track width 10mm	2mm of plasma observed Lower level audible noise present Large quantity of small sparks present in all directions up to 100mm from target Powder was pushed to one side Track width 7mm NO SHIELD GAS	Same level of audible noise as previous run Intermittent sparking, 50mm from target in all directions Imm of white plasma observed Track width 5mm
-	2 Process Tv	: 114.3 mm	Travel Distance (mm)	25.4	25.4	25.4	25.4
} ss	Date: 11-01-2000 Test Coupon(s): 2 Material(s): Black Painted Ti-6Al-4V Substrate & Al-2219 PPP	Focal Position: 114.3 mm	Process Speed (mm/min)	12.7	12.7	12.7	12.7
Table 6s	1-2000 T	0 SCFH	Beam Mode (kHz)	10	10	20	20
)	Date: 11-01-2000	Gas: He @ 10 SCFH	Voltage (V)	10.73	10.73	10.73	10.73
-	L Painted 7	Shield G	Ave. Laser Power (W)	~35	~35	~35	~35
	erial(c). Rlac	(3): Time	Diode Current (A)	46.3	46.3	46.3	46.3
	Mak		Run Number	1	2	3	4

	Material(s): Black Painted Ti-6Al-4V Substrate & Al-2219 PPP Process Type: PPP Cladding		Comments	No noise, sparks, or smoke observed Plasma <1 mm high observed Track width < 5 mm	No noise, sparks, or smoke observed Plasma <1mm high observed Track width < 5mm HAZ much narrower than previous runs
2	Process Tyl	114.3 mm	Travel Distance (mm)	25.4	25.4
Date: 11-01-2000 Test Coupon(s): 2	AI-2219 PPP	Shield Gas: He @ 10 SCFH Focal Position: 114.3 mm	Process Speed (mm/min)	12.7	12.7
1-2000 T	ubstrate &	0 SCFH	Beam Mode (kHz)	50	CW
Date: 11-0]	Ti-6Al-4V S	3as: He @ 10	Voltage (V)	10.73	10.73
	k Painted	Shield (Ave. Laser Power (W)	~35	41.4
	erial(s): Blac		Diode Current (A)	46.3	46.3
	Mat		Run Number	ĸ	9

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Table 6t

Test Coupon(s): 2	Material(s): Black Painted Ti-6Al-4V Substrate & Al-2219 PPP Process Type: PPP Cladding	He @ 10 SCFH Focal Position: 114.3 mm	HAZ HAZ Comments (µm) (µm)	607.2				98.0		
Date: 11-01-2000 Test Coupon(s): 2	i-6Al-4V Substrate	as: He @ 10 SCFH	Clad Height HAZ HAZ Clad Height Width Depth (µm) (µm) (µm)	790.4		390.8		346.4	50.8	
	ck Painted T	Shield Gas:	Clad Depth (µm)	245.2	366.4	232.8	296.4	48.8	44.8	
	terial(s): Bla		Clad Width (µm)	525.6	336.4	772.0	485.2	346.4	377.6	
	M:		Run Number	1+	2+	3+	4+	5+	+9	

⁺ Measurements obtained from optical microscopy

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			Date: 11	Date: 11-01-2000 T	Test Coupon(s): 3	3		
	Materia	Material(s): Black Painted Al-2219 Substrate & Ti-6Al-4V PPP	Al-2219 S	ubstrate & T	i-6Al-4V PPP	Process T	ype: PPF	Process Type: PPP Cladding
		Shield Gas:	Gas: He @	He @ 10 SCFH	Focal Position: 114.3 mm	114.3 mm		
Run Number	Diode Current (A)	Ave. Laser Power (W)	Voltage (V)		Beam Mode Process Speed (kHz) (mm/min)	Travel Distance (mm)	Clad Width (µm)	Comments
1>	46.3	~35	10.73	10	12.7	25.4	889	High pitched noise present 1mm white plasma observed Large sparks observed 30mm from target in all directions NO SHIELD GAS
2 >>	46.3	~35	10.73	10	12.7	25.4	8/9	High pitched noise present 1 mm white plasma observed Small sparks observed 10mm from target in all directions 7mm track width
.	46.3	~35	10.73	20	12.7	25.4	781	Lower level of audible noise present Intermittent sparking observed in direction of gas
4	46.3	~35	10.73	50	12.7	25.4	1398	No noise or sparks observed <1mm yellow plasma observed
5^	46.3	41.4	10.73	CW	12.7	25.4	1234	No noise or sparks observed <1mm yellow plasma observed
A Mos	curomonte modo	A Moognusoments made by abotecrophic english	nalveic					

^ Measurements made by photographic analysis

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Date: 11-01-2000 Test Coupon(s): 4 Material(s): Untreated Ti-6Al-4V Process Type: Nitridation Shield Gas: N2 @ 10 SCFH Focal Position: 114.3 mm	1* 385.2 173.8 318.6 505.5 61.2	7* 254.3 438.7 607.5 594.6
---	---------------------------------	----------------------------

^{*} Measurements obtained from scanning electron microscopy

Table 6w

				I abic on			
			Date: 11-01-2000		Test Coupon(s): 5	5	
Ma	terial(s): Bla	ck Paintec	i Al-2219 Sub	strate &	Material(s): Black Painted Al-2219 Substrate & Al-2219 PPP	Process Type:	Process Type: PPP Cladding
		Shield Gas:	Gas: He @ 10 SCFH		Focal Position: 114.3 mm	114.3 mm	
Run Number	Diode Current (A)	Ave. Laser Power (W)	Voltage (V)	Beam Mode (kHz)	Process Speed (mm/min)	Travel Distance (mm)	Comments
-	46.3	41.4	10.73	CW	12.7	25.4	No plasma, sparking or smoke observed
7	46.3	~35	10.73	20	12.7	25.4	No plasma, sparking or smoke observed
3	46.3	-35	10.73	10	12.7	25.4	NO SHIELD GAS Audible noise present Blue plasma observed Shield gas turned on halfway thru run
4	46.3	~35	10.73	15	12.7	25.4	Less noise than previous run <1 mm white plasma observed
5	46.3	~35	10.73	20	12.7	25.4	No plasma, sparking or smoke observed

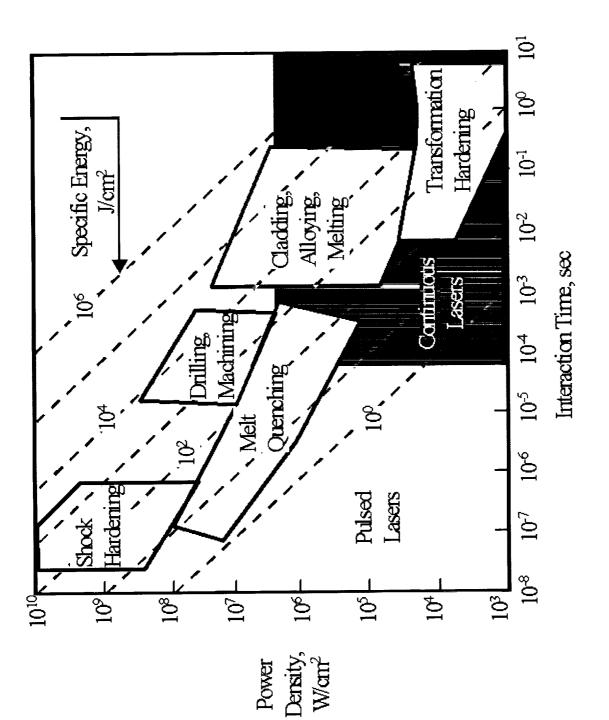
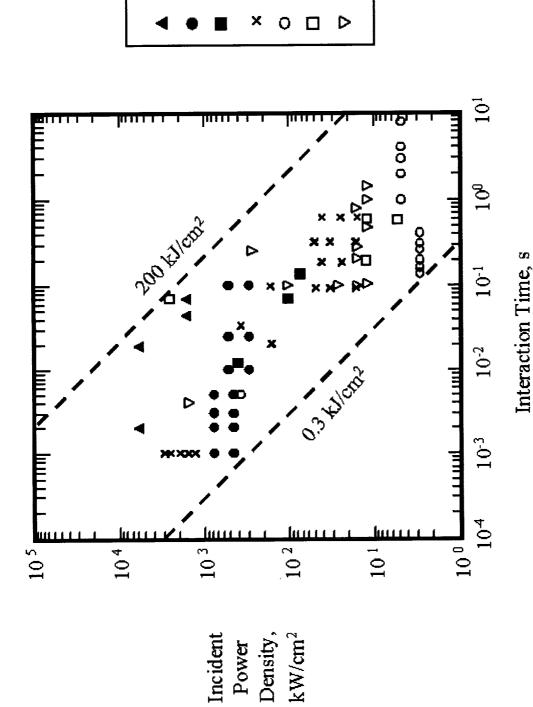


Figure 1: Laser beam power density, specific energy and interaction times for various regimes



Aluminum alloys

Zirconium alloys

Titanium alloys

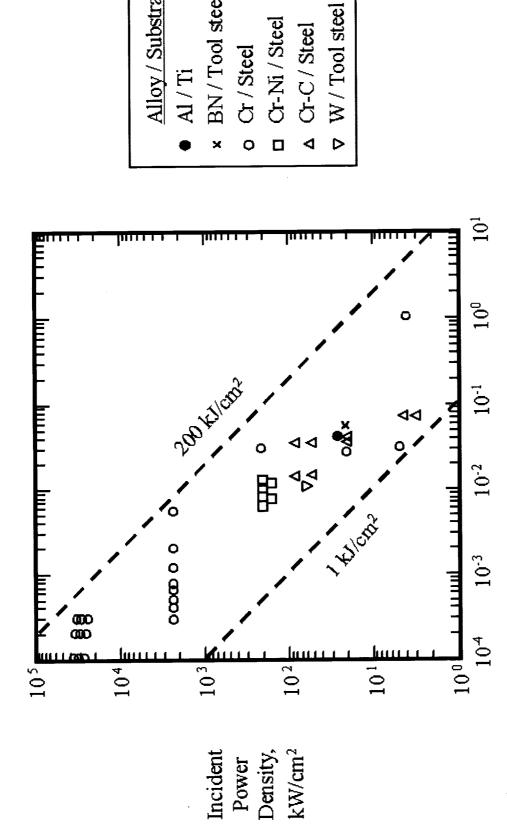
Low alloy steels

Stainless steels

Cast irons

Tool steels

Figure 2: Processing map for laser surface melt quenching



Alloy / Substrate Al / Ti BN / Tool steel Cr / Steel

Figure 3: Processing map for laser surface alloying

Interaction Time, s

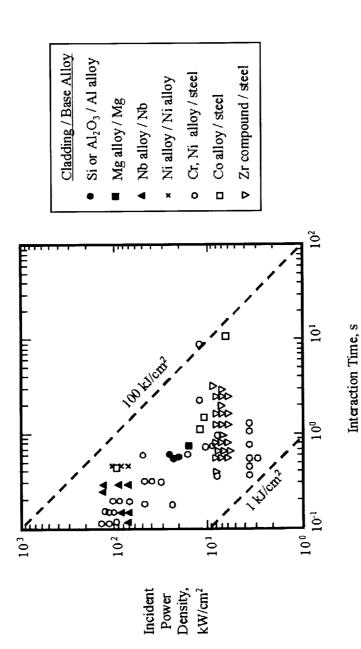
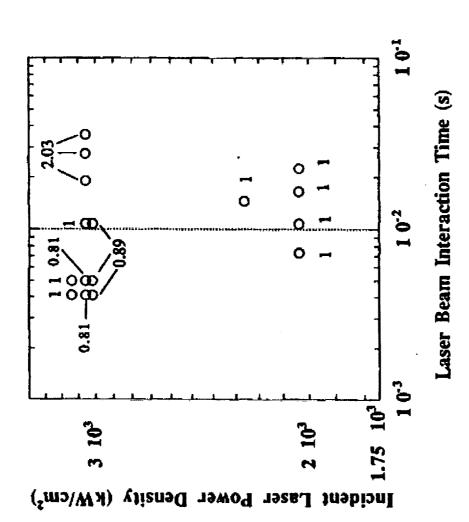


Figure 4: Processing map for laser surface cladding



All dimensions in the plot refer to weld depth in mm.

Figure 5a: CO₂ Laser Welding Data for Ti-6Al-4V

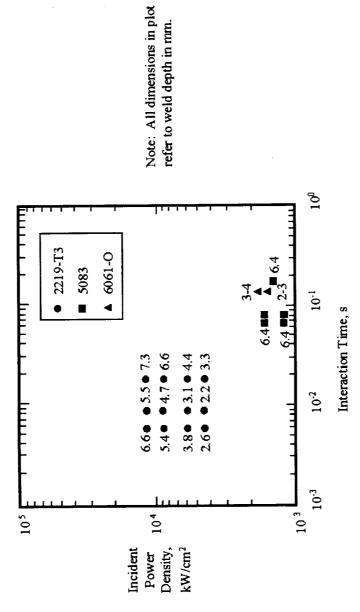
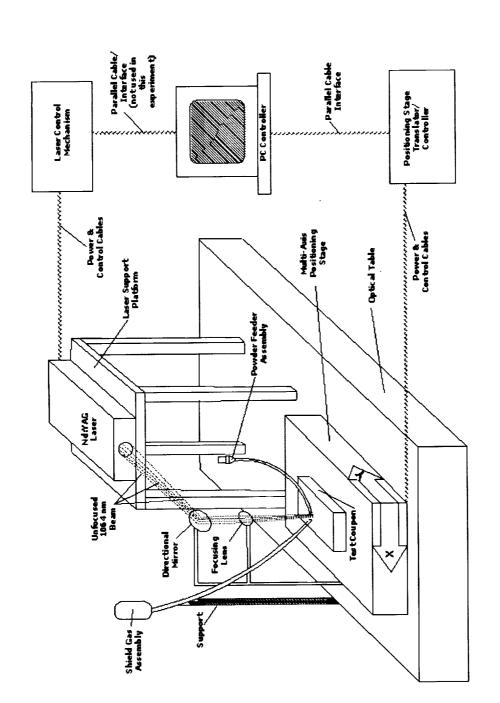


Figure 5b: Processing map for laser welding of aluminum alloys



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Figure 6a: Experimental Setup Overview

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Figure 6b: Experimental Arrangement

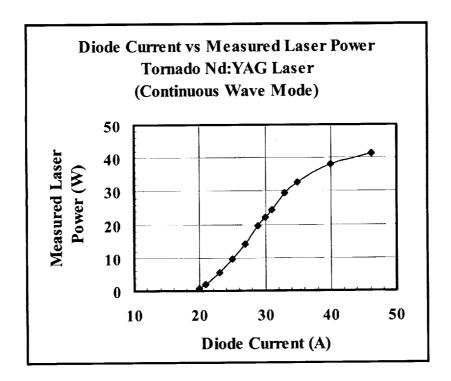


Figure 7a: Laser Power vs. Diode Current Characteristics (Continuous Wave Mode)

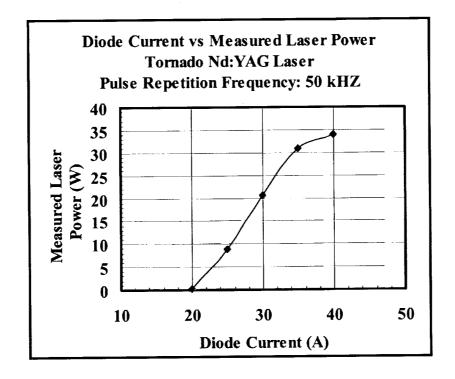


Figure 7b: Laser Power vs. Diode Current Characteristics (Pulsed Mode with PRF = 50 kHz

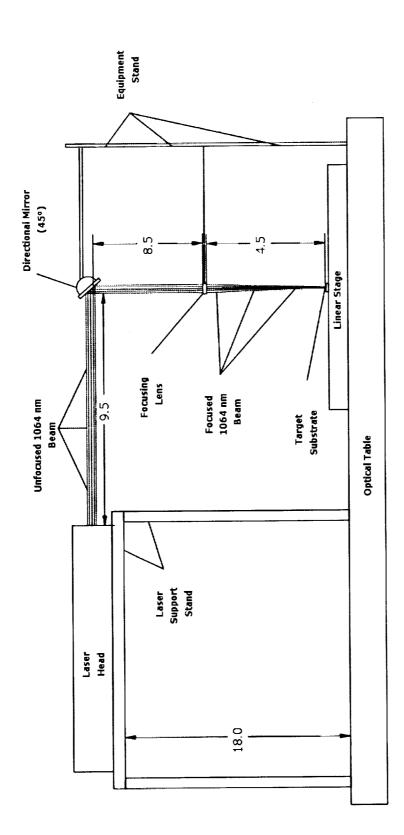


Figure 8: Laser Beam Directional Diagram (All Dimensions in Inches--Not Drawn to Scale)

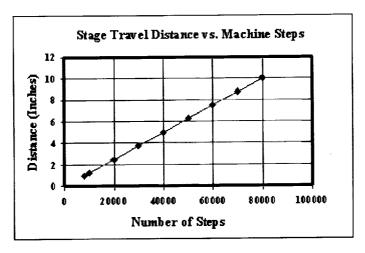


Figure 9a: Stage Travel Distance vs Number of Machine Steps Traveled

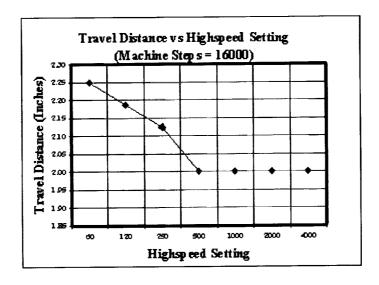


Figure 9b: Travel Distance at Various Machine Highspeed Settings

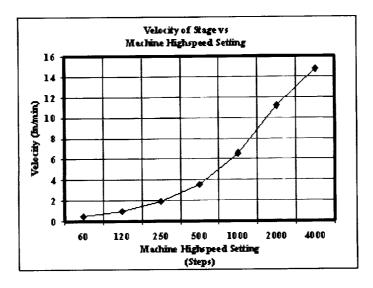


Figure 9c: Stage Velocity at Various Machine Highspeed Settings

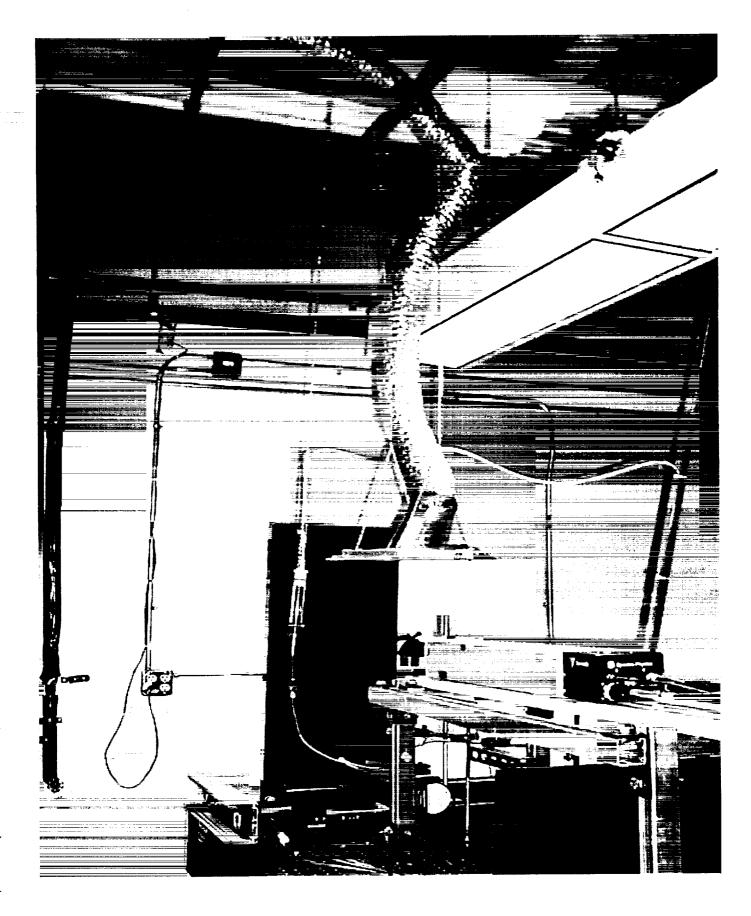


Figure 10: Ventilation System

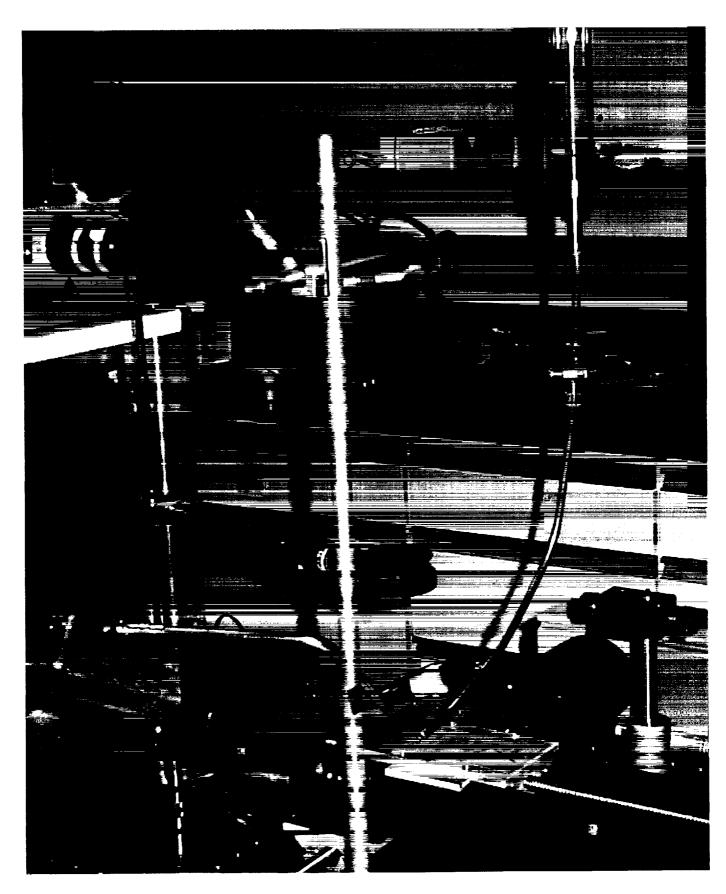


Figure 11: Powder Delivery System

Figure 12a: As-Received Ti-6Al-4V (2660x)

Figure 12b: As-Received Al-2219 (519x)

Figure 12c: As-Received Al-2219 (2660x)

Figure 12d: As-Received Al-2090 (2660x)

Figure 12e: As-Received Al-2090 (5220x)

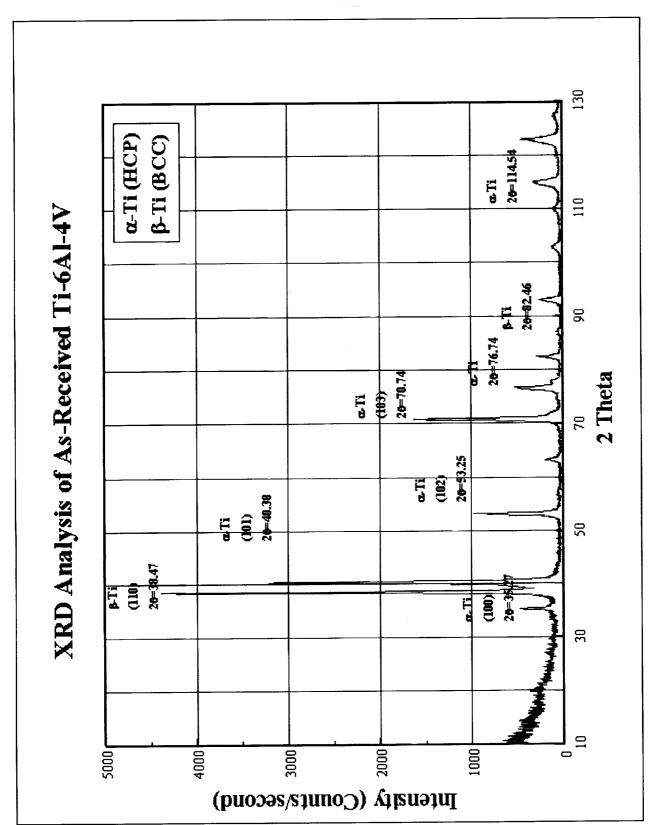


Figure 13a: XRD Scan of As Received Ti-6Al-4V

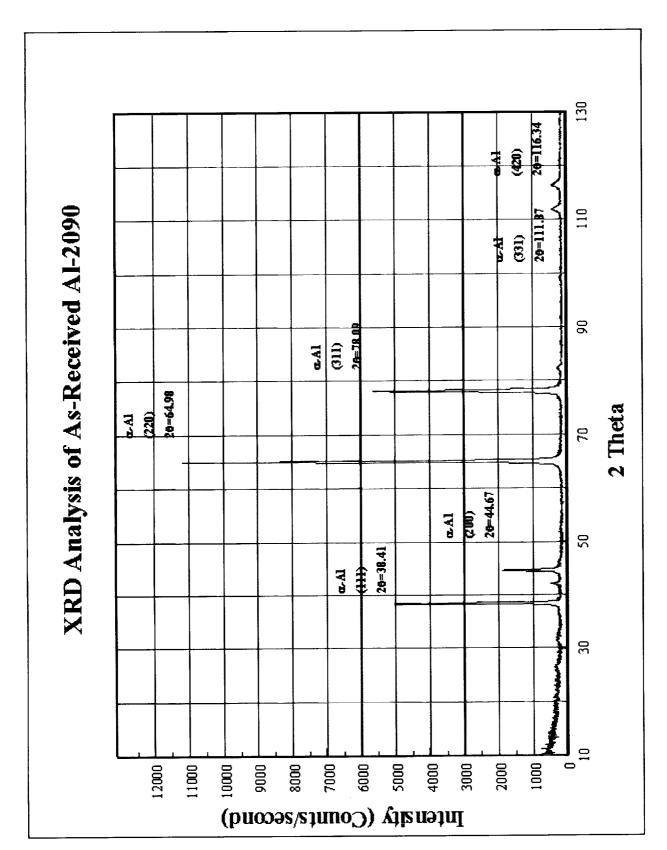


Figure 13b: XRD Scan of As Received Al-2090

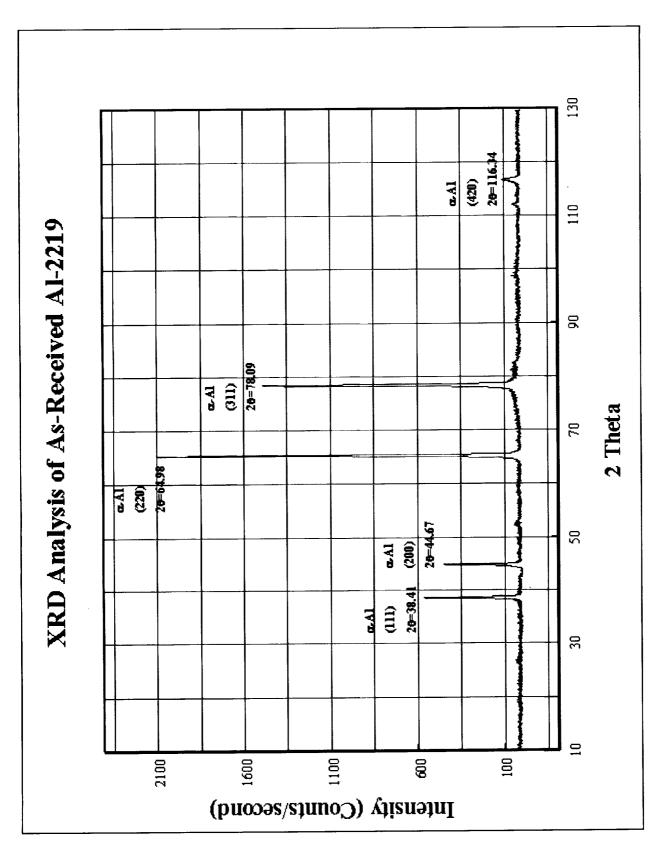


Figure 13c: XRD Scan of As Received Al-2219

Figure 14a: Laser Processed Ti-6Al-4V (10-3-2000 TC1-4)

Figure 14b: Laser Processed Ti-6Al-4V (After Partial Chemical Etching) (10-9-2000 TC2)

Figure 14c: Laser Processed Al-2219 (10-9-2000 TC3)

Figure 14d: Laser Processed Ti-6Al-4V (10-13-2000 TC1)

Figure 14e: Black Painted Al-2090 (10-27-2000 TC1)

Figure 14f: Laser Processed Al-2090 (10-27-2000 TC2)

Figure 14g: Laser Processed Al-2219 (10-30-2000 TC3)

Figure 14h: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (10-30-2000 TC4)

Figure 14i: Black Painted Al-2090 Substrate & Ti-6Al-4V PPP (11-1-2000 TC1)

Figure 14j: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (11-1-2000 TC2)



Figure 14k: Black Painted Al-2219 Substrate & Ti-6Al-4V PPP (11-1-2000 TC3)

Figure 141: Laser Nitrided Ti-6Al-4V (11-1-2000 TC4)

Figure 15a: Laser Cutting on Untreated Ti-6Al-4V (PRF = 1 kHz) (10-03-2000 TC4 R1 20x)

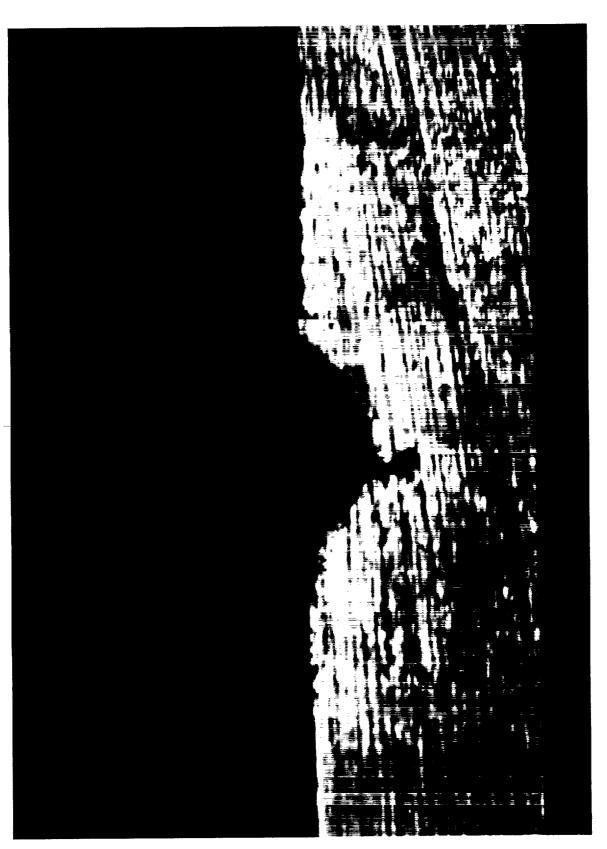


Figure 15b: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP Pulsed Mode; PRF: 20 kHz; Speed:12.7 mm/min (11-1-2000 TC2 R3 20x)

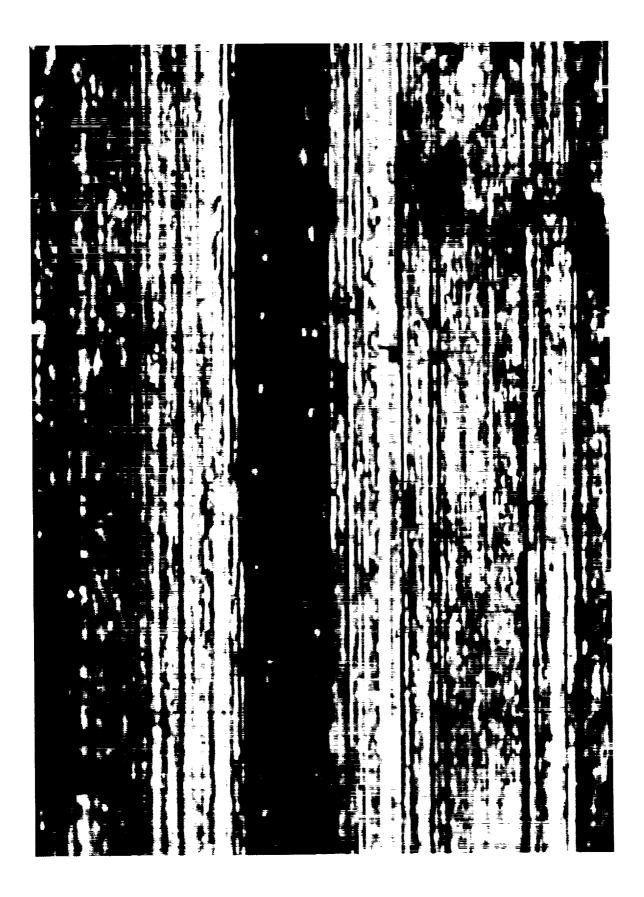


Figure 15c: Black Painted Al-2219 (10-30-2000 TC2 R2 05x Top)

Figure 15d: Black Painted Al-2219 (10-30-2000 TC2 R2 20x Top)

Figure 15e: Black Painted Al-2219 (10-30-2000 TC2 R3 20x Top)

Figure 15f: Untreated Al-2219 (10-30-2000 TC3 R2 20x)

Figure 15g: Untreated Al-2219 (10-30-2000 TC3 R3 20x Top)

Figure 15h: Untreated Al-2219 (10-30-2000 TC3 R4 5x)

Figure 15i: Black Painted Al-2090 (Top Surface; Pulsed Mode-PRF: 20 kHz; Speed: 12.7 mm/min) (10-27-2000 TC1 R4 10x Top)

Figure 15j: Black Painted Al-2090 (10-27-2000 TC1 R4 20x Top)



Figure 15k:Black Painted Al-2090 ()
(Top Surface; Pulsed Mode-PRF: 1 kHz; Speed: 12.7 mm/min)
(10-27-2000 TC1 R5 10x Top)

Figure 15l: Black Painted Al-2090 (Transverse Section) (10-27-2000 TC1 R5 10x)

Figure 15m: Black Painted Al-2090 (10-27-2000 TC1 R5 20x Top)

Figure 16a: Untreated Ti-6Al-4V Substrate and Ti-6Al-4V PPP (cw mode; Power: 46.3 W; Speed: 25.4 mm/min) (10-10-2000 TC1 R3 0450x)

Figure 16b: Untreated Ti-6Al-4V Substrate and Ti-6Al-4V PPP (Interface) (10-10-2000 TC1 R3 2890x)

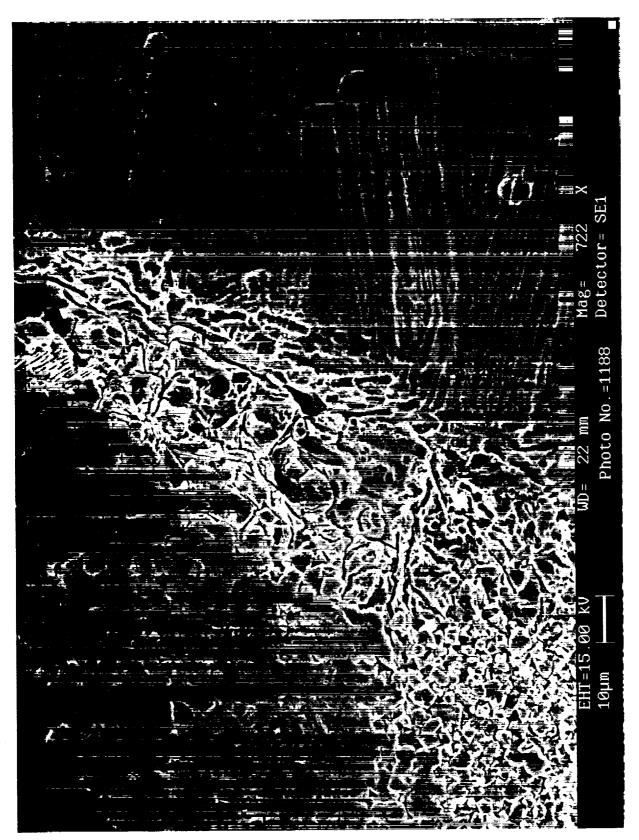


Figure 16c: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (10-30-2000 TC4 R5 0637x (Top Center of Coating))

Figure 16d: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (10-30-2000 TC4 R5 88x)

Figure 16e: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (10-30-2000 TC4 R5 0637x (Top Center of Coating))

Figure 16f: Black Painted Ti-6Al-4V Substrate & Al-2219 PPP (10-30-2000 TC4 R5 0637x SC Interface)

Figure 16g: Laser Nitridation of Untreated Ti-6Al-4V (11-1-2000 TC4 R1 88x)

Figure 16h: Laser Nitridation of Untreated Ti-6Al-4V(Interface) (11-1-2000 TC4 R1 2660x)

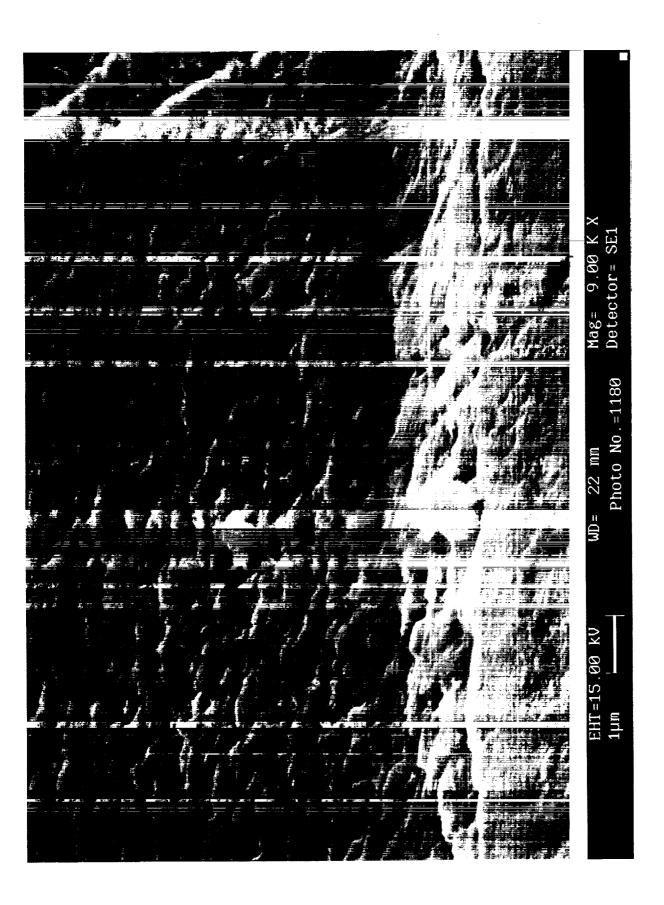


Figure 16i: Laser Nitridation of Untreated Ti-6Al-4V(Top Surface) (11-1-2000 TC4 R1 9000x)

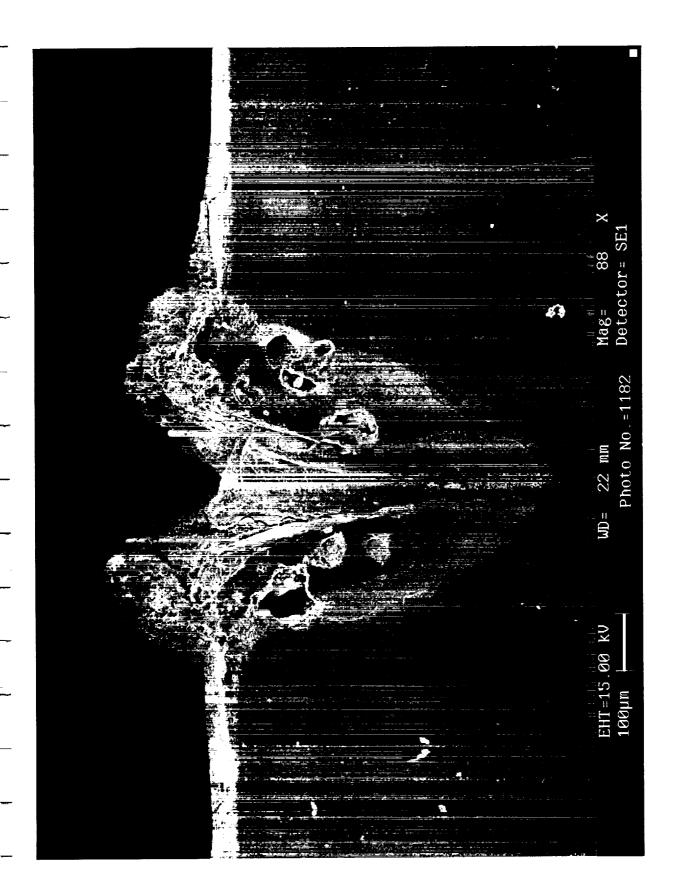


Figure 16j: Laser Nitridation of Untreated Ti-6Al-4V(PRF = 10 kHz) (11-1-2000 TC4 R7 88x)

Figure 16k: Untreated Al-2219 (10-30-2000 TC2 R3 325x)

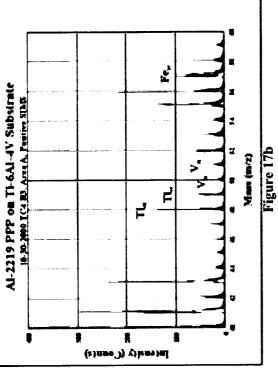
Figure 16l: Untreated Al-2219 (10-30-2000 TC3 R6 172x)

Figure 16m: Untreated Al-2219 (10-30-2000 TC3 R10 172x)

Figure 16n: Untreated Al-2219 (10-30-2000 TC3 R10 1630x)

Figure 160: Black Painted Al-2090 (Top Surface) (Pulsed Mode; PRF: 20 kHz; Speed: 12.7 mm/min) (10-27-2000 TC1 R4 200x)

Figure 16p: Black Painted Al-2090 (10-27-2000 TC1 R4 3000x)





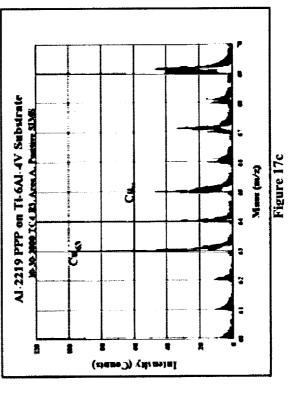
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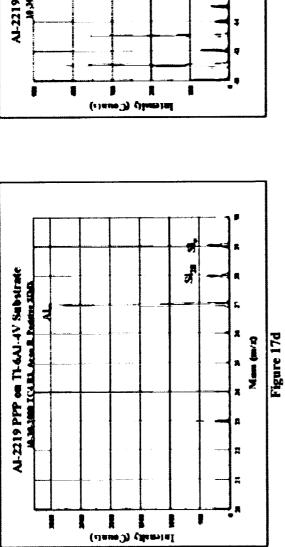
. . . .

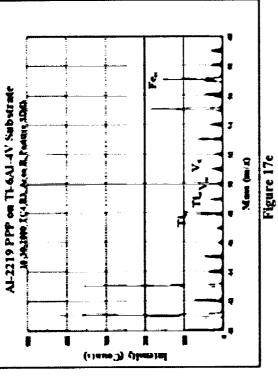
Intendity (County)

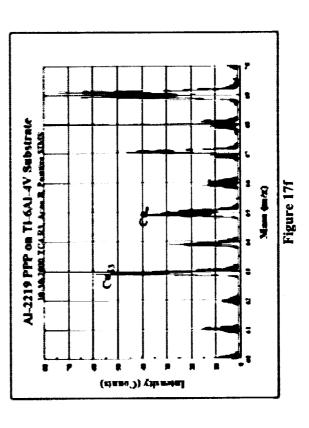
AI-2219 PPP on TI-6AI-4V Substrute
18-30-300 ICA 31, Anna A. Panier, SIMB.



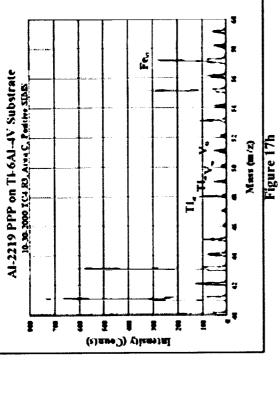
Figures 17a-c: SIMS Analysis of Al-2219 PPP on Ti-6Al-4V Substrate

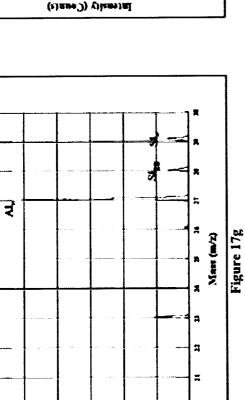






Figures 17d-f: SIMS Analysis of Al-2219 PPP on Ti-6Al-4V Substrate





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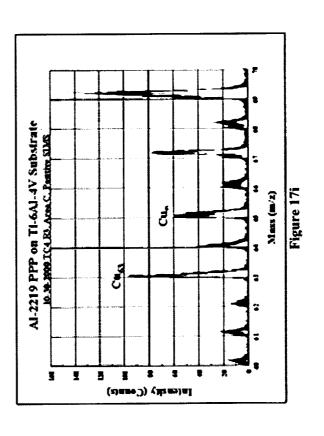
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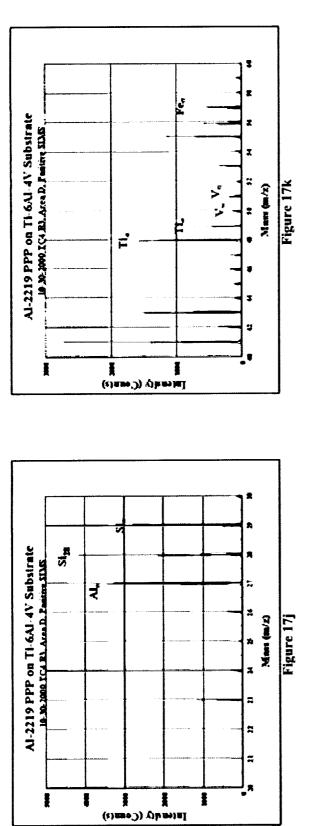
Al-2219 PPP on Ti 6Al 4V Substrate

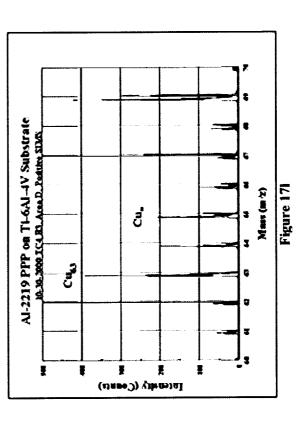
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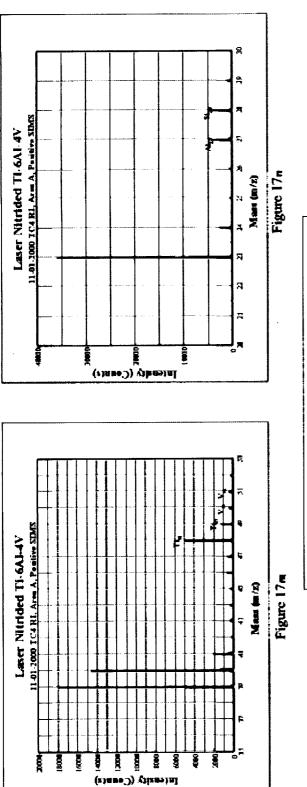


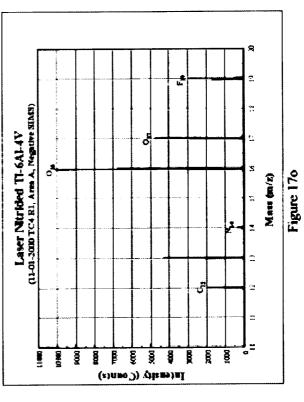
Figures 17g-i: SIMS Analysis of Al-2219 PPP on Ti-6Al-4V Substrate





Figures 17j-l: SIMS Analysis of Al-2219 PPP on Ti-6Al-4V Substrate





Figures 17m-o: SIMS Analysis of Al-2219 PPP on Ti-6Al-4V Substrate

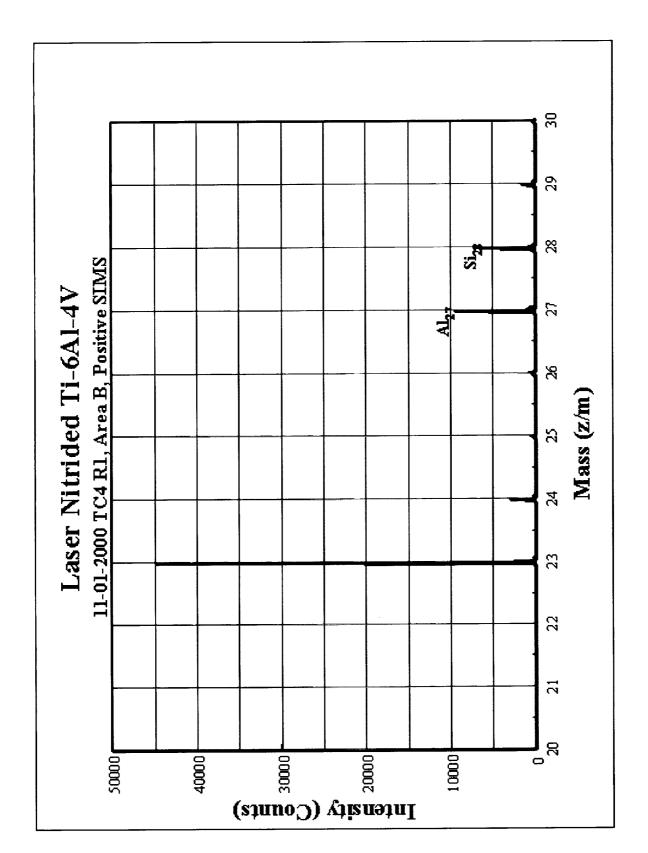


Figure 17p: Laser Nitrided Ti-6Al-4V

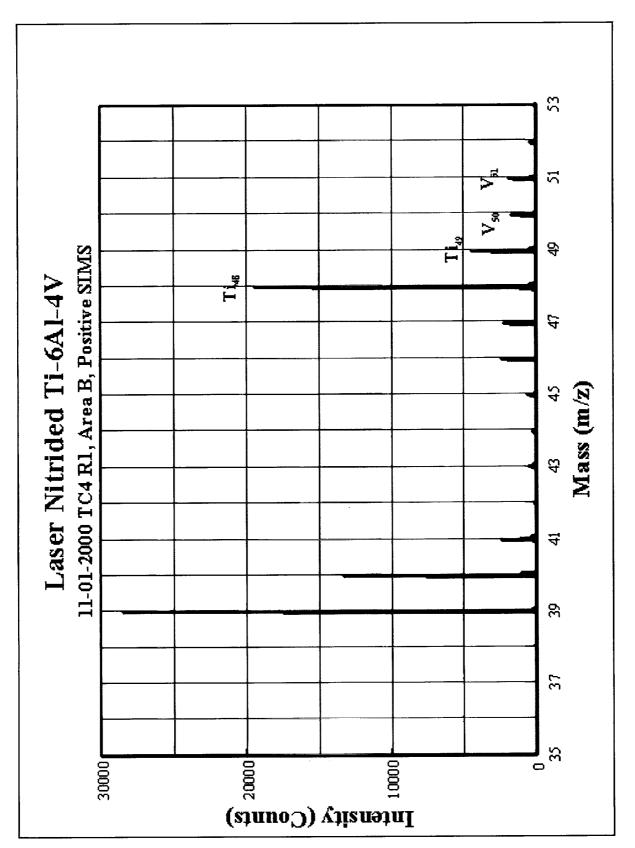


Figure 17q: Laser Nitrided Ti-6Al-4V

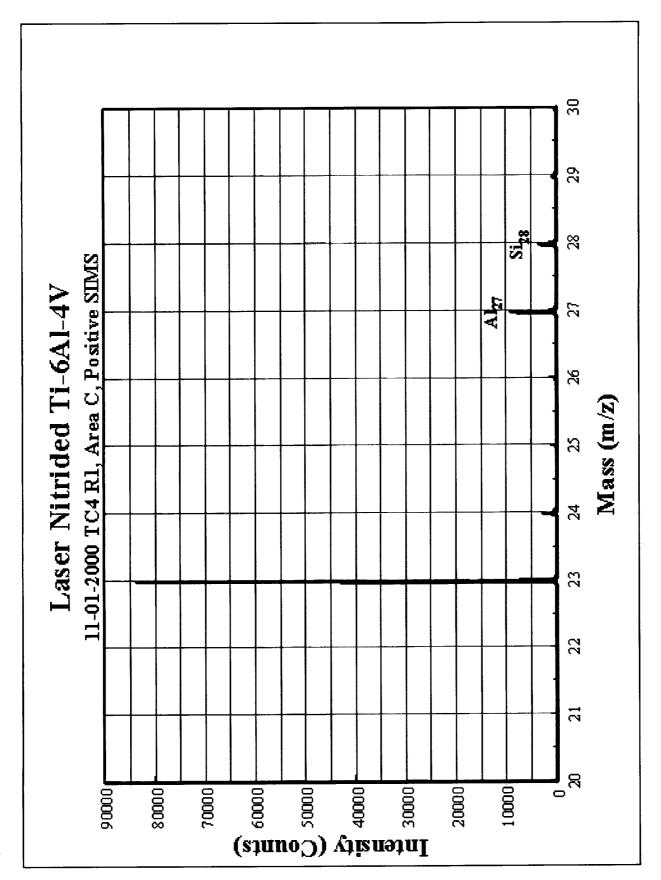


Figure 17r: Laser Nitrided Ti-6Al-4V

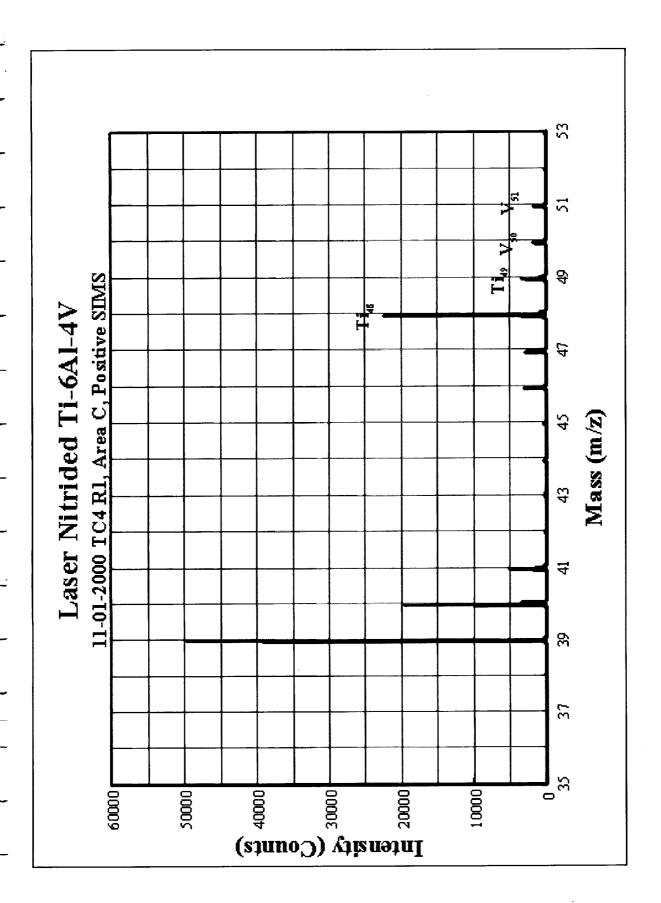


Figure 17s: Laser Nitrided Ti-6Al-4V

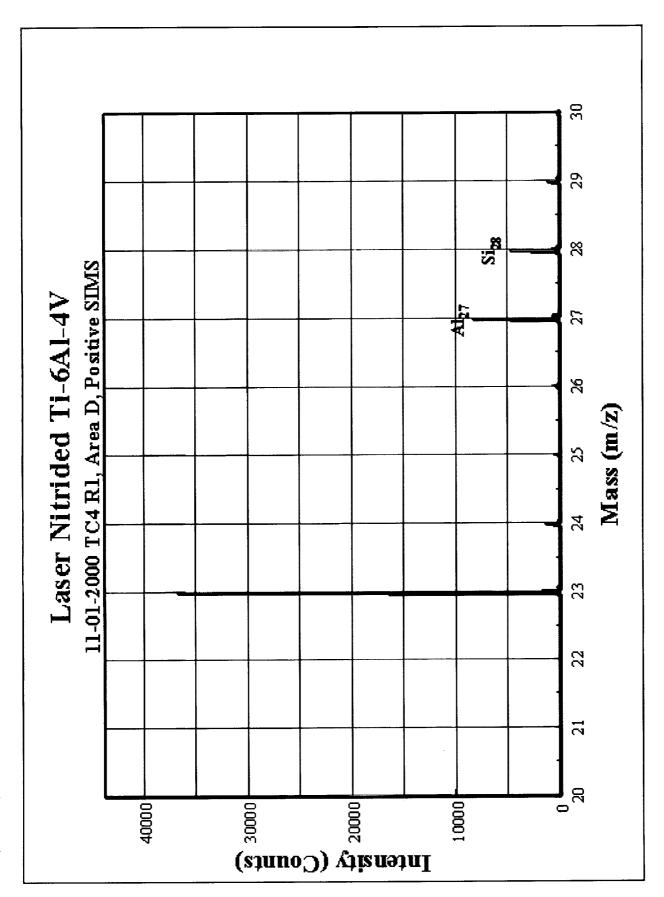


Figure 17t: Laser Nitrided Ti-6Al-4V

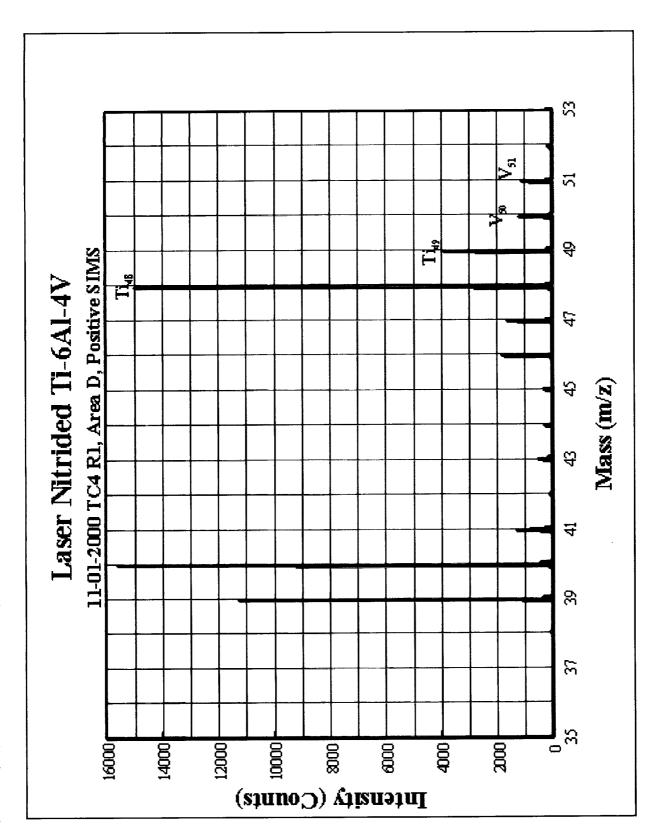


Figure 17u: Laser Nitrided Ti-6Al-4V

Figure 18a: SEM Cross-Section of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18b: EDS Analysis of Entire Cross-Section of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18c: SEM Cross-Section of Laser Nitrided Ti-6Al-4V Showing Location of Selected Analysis Point Sample 11-01-2000 TC4 R1)

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Figure 18d: EDS Analysis of Selected Area of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

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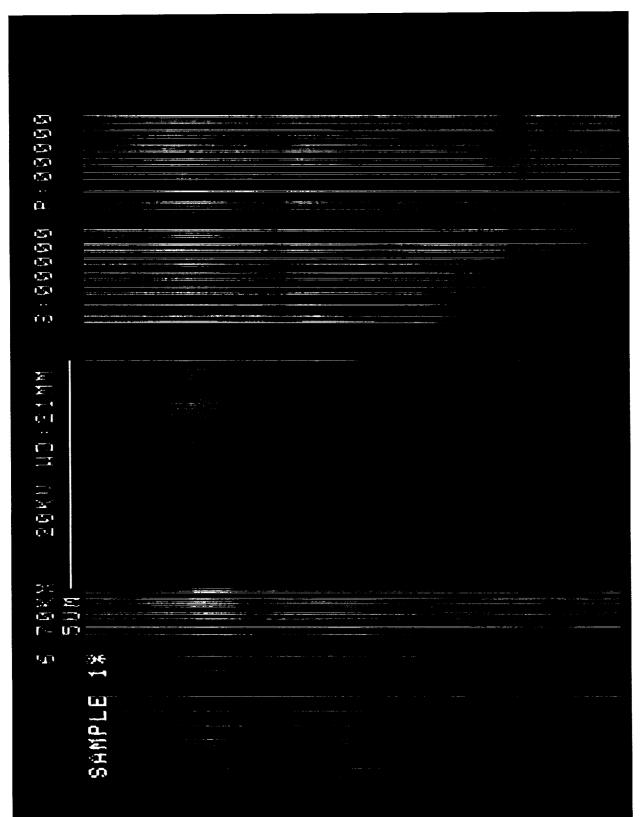


Figure 18e: High Magnification SEM Micrograph of Selected Analysis Point of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18f: SEM Cross-Section of Laser Nitrided Ti-6Al-4V Showing Location of Selected Analysis Point (Sample 11-01-2000 TC4 R1)

Figure 18g: EDS Analysis of Selected Area of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18h: High Magnification SEM Micrograph of Selected Analysis Point of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18i: SEM Cross-Section of Laser Nitrided Ti-6Al-4V Showing Location of Selected Analysis Point (Sample 11-01-2000 TC4 R1)

Figure 18j: EDS Analysis of Selected Area of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18k: High Magnification SEM Micrograph of Selected Analysis Point of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 181: SEM Cross-Section of Laser Nitrided Ti-6Al-4V Showing Location of Selected Analysis Point (Sample 11-01-2000 TC4 R1)

Figure 18m: EDS Analysis of Selected Area of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 18n: High Magnification SEM Micrograph of Selected Analysis Point of Laser Nitrided Ti-6Al-4V (Sample 11-01-2000 TC4 R1)

Figure 180: SEM Cross-Section of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

Figure 18p: EDS Analysis of Entire Cross-Section of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

Figure 18q: SEM Cross-Section of Al-2219 PPP on Ti-6Al-4V Substrate Showing Location of Selected Analysis Point (Sample 10-30-2000 TC4 R3)

Figure 18r: EDS Analysis of Selected Area of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

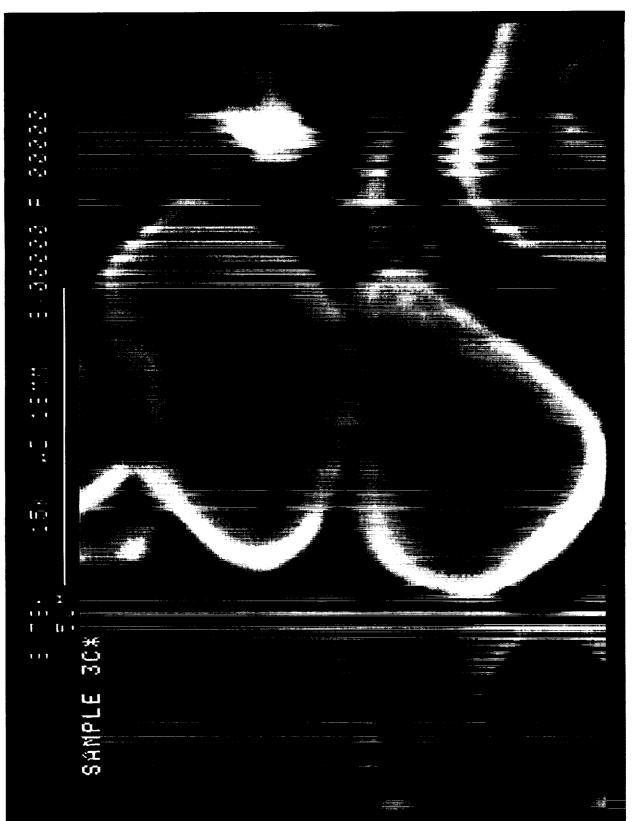


Figure 18s: High Magnification SEM Micrograph of Selected Analysis Point of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

Figure 18t: SEM Cross-Section of Al-2219 PPP on Ti-6Al-4V Substrate Showing Location of Selected Analysis Point (Sample 10-30-2000 TC4 R3)

Figure 18u: EDS Analysis of Selected Area of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

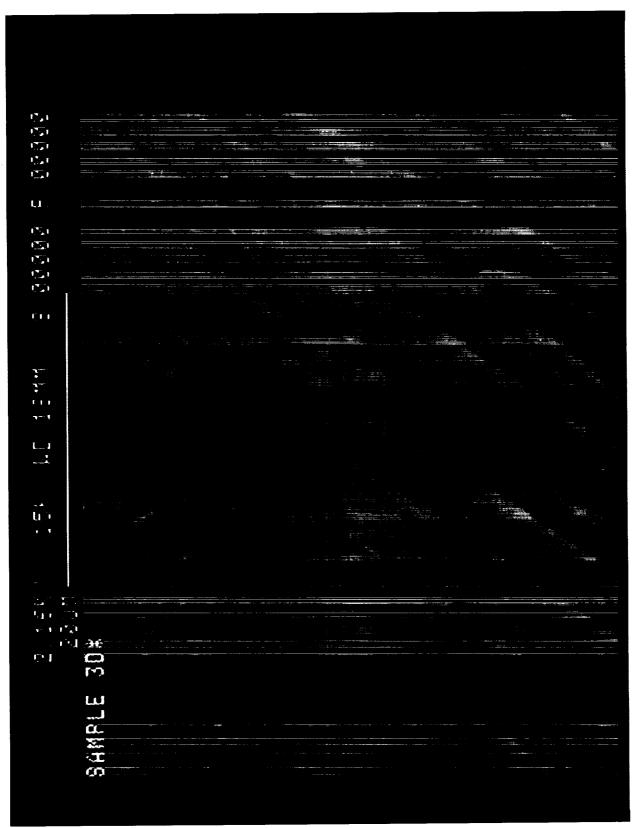
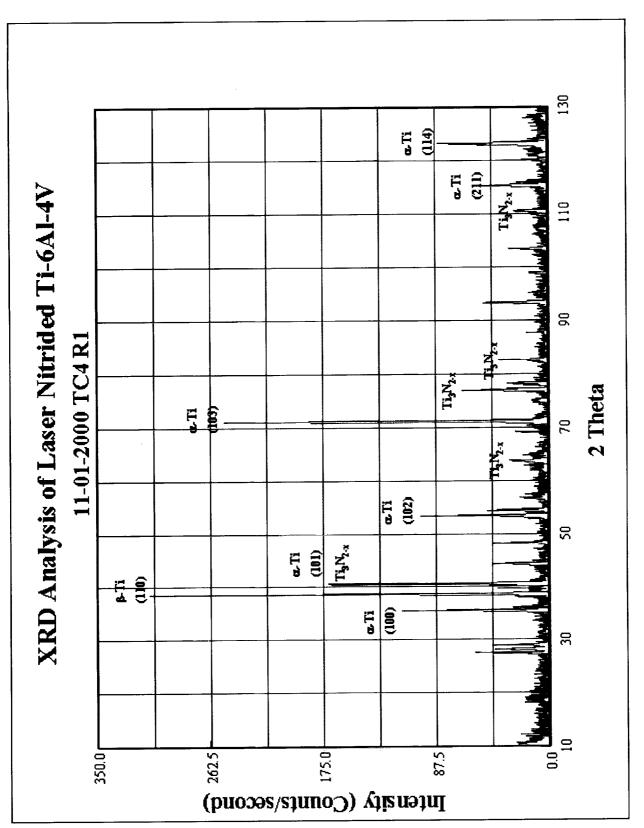


Figure 18v: High Magnification SEM Micrograph of Selected Analysis Point of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

Figure 18w: SEM Cross-Section of Al-2219 PPP on Ti-6Al-4V Substrate Showing Location of Selected Analysis Point (Sample 10-30-2000 TC4 R3)

Figure 18x: EDS Analysis of Selected Area of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)

Figure 18y: High Magnification SEM Micrograph of Selected Analysis Point of Al-2219 PPP on Ti-6Al-4V Substrate (Sample 10-30-2000 TC4 R3)



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Figure 19a: XRD Analysis of Laser Nitrided Ti-6Al-4V (11-01-2000 TC4 R1)

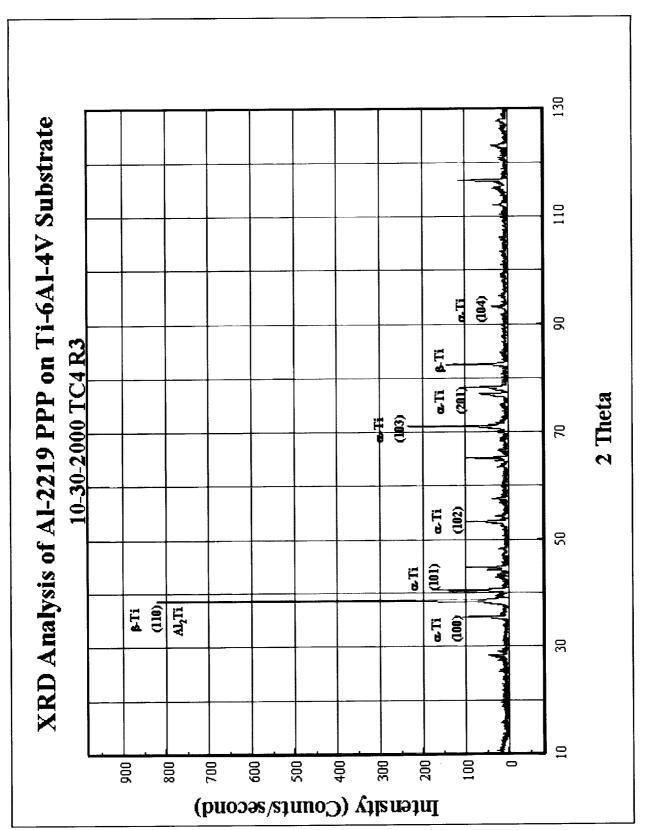


Figure 19b: XRD Analysis of Al-2219 PPP on Ti-6Al-4V Substrate (10-3002000 TC4 R3)

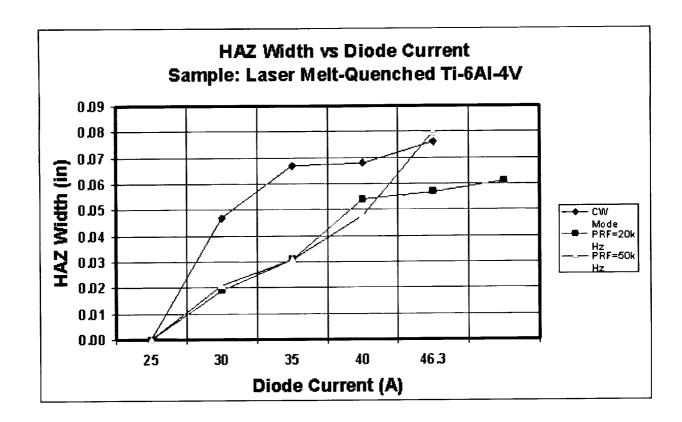


Figure 20a: HAZ Width vs. Diode Current

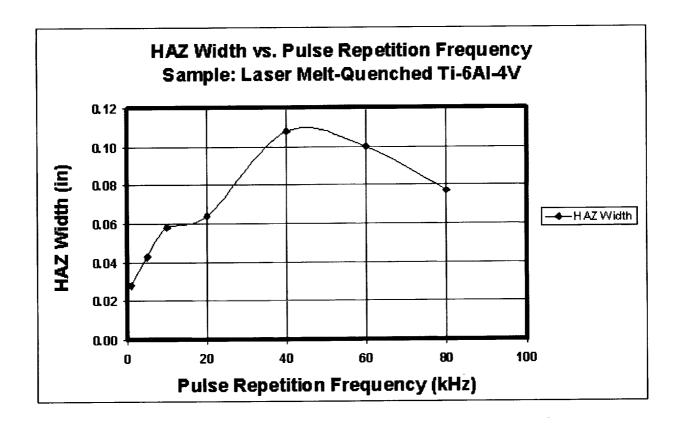


Figure 20b: HAZ Width vs. Pulse Repetition Frequency

APPENDIX A: CONTINUOUS WAVE CO, LASER BEAM WELDING DATA FOR AL-ALLOYS

MATERIAL: A-ALLOYS
LASEN BEAM TYPE: COMMUNISWAYE

	ļ	Average	1	-	į			1		
	•	į	Otemeter	Į	Denet	Energy	į	Gas & Flore		j.
Speed to other	Pers Stade	(W.W.)	(86)	(4/44)	(hW/cm+2)	(B.4/em.2)	101	Rote (m-3/hr)	Chereteriettee	
A5052P-H32	8	1,600		18.33				Ψť		Manumato, et al, 1992
t-tem to 6mm	8	1,800		108.33				Ar		Magumato, et al, 1992
A7075P-T6	8							¥		Masumato, et al., 1992
5063	8	6.000		100.00						Yoshikawa, M., et al, 1905
tellers to Smen									BW & LW	
5251	300	6.000		100.00					BW & LW	Yoshikama, M., et al., 1995
In the bring										
1908	ğ	5.000		100.00					BW & LW	Yoshikawa, M., et al., 1995
t-Imm to 3mm	8	8.000		100.00					BW & LW	Yoshikana, M., et al, 1995
6082	8	8,000		100,00					BW & LW	Yoshikawa, M., et al., 1995
telms to Smm	8	5.000		100.00					8W & LW	Yoshikawa, M., et al., 1985
Al 8251, 1-2mm	8	9.000		100.00					ANE	Richee, S.T., 1993
AJ 6061, 1-2mm	8	8.000		100.00					BW	Richee, S.T., 1963
A2219-T3	8	3,000	0.291	16.67	4499.88	61,78	0.017481	Ar, 2.41	8	Kutsune, M., et al., 1983
ta-Amen & 10mm	8	3.000	0.291	33.33	4499.88	30.89	0.0087405	Ar, 2.41	8	Kuteune, M., et al, 1993
	8	3.000	0.291	50.00	4499.88	20.59	0.005827	Ar, 2.41	8	Kutsuna, M., et at, 1993
	8	4.000	0.201	18.67	5999.85	82.38	0.017481	A, 2.41	å	Kutsuna, M., et al, 1993
	8	4.000	0.291	33.33	5989,85	41.19	0.0087405	A, 2.41	8	Kuteuma, M., et al, 1993
	8	4.000	0.291	80.00	5969.85	27.46	0.005827	A, 2.41	8	Kutsune, M., et al, 1993
	8	6.000	0.201	16.67	6969.77	123.56	0.017481	Ar, 2.41	e S	Kutsuna, M., et al., 1993
	8	6.000	0.201	33,33	8999.77	61.78	0.0087405	A. 2.41	ŝ	Kutsuns, M., et al., 1993
	8	6.000	0.291	50.00	\$999.77	41.10	0.005827	Ar. 2.41	ŝ	Kutsuna, M., et al., 1993
	8	8.000	0.291	16.67	11999.60	164,75	0.017481	Ar. 2.41	2	Kutaura, M., et al. 1993
	8	9.000	0.291	33.33	11998.68	82.38	0.0087405	A, 2.41	ŝ	Kuteure, M., et el, 1983
	200	9.000	0.291	\$0.00	11998.68	54,92	0.005827	A, 2.41	8	Kutsura, M., et al., 1993
A6052	8		0.291	16.67			0.017481	A. 2.41	B	Kutsuna, M., et al., 1993
ASOBS	800	1.900	0.291	3.33	2849.93	195.64	0.087405	He, 1.19	ĝ	Kuteune, M., et al., 1993
	g	1.900	0.291	9.00	2849,93	130.43	0.05627	He, 1.19	8	Kutsune, M., et el., 1993
	8	1.900	0,291	8.33	2840.93	78.26	0.034962	He, 1,19	æ	Kutsurs, M., et el., 1983
A6083P-O	8	1.900	0.291	2.8	2849.93	130.43	0.05827	He, 1.51	8	Kutsume, M., et ef, 1983
1-80x40x6 (mm) &		1.900	0.291	9.00	2849.93	130.43	0.05827	He, 0.9 &	80	Kutsune, M., et al., 1993
1-BOX 40x4 (MM)	8		0.291					N2, 0.612	8	Kutsune, M., et el, 1983
A6063	COZ @ TEMOS	1.000		50.00				A. 2.90	BOP & IBW	Kutsune, M., et el, 1993
	CO2 @ TEMO!	2.000		\$0.00				Ar, 2.99	BOP & IBW	Kutsune, M., et al., 1993
	CO2 @ TEMO1 2.500	2.500		\$0.00				A, 2.99	BOP & BW	Kutsune, M., et al., 1993
	CC2 @ TEM01	3,100		\$0.00				Ar, 2.99	BOP & IBW	Kutsuna, M., et al., 1993
	COZ @ TEMO!	3.500		50.00				Ar, 2.99	BOP & BW	Kultsume, M., et al., 1993
	CO2 @ TEMO1			50.00				Ar. 2.90	BOP & BW	Kuleune, M., et al. 1903
	CC2 @ TEM01	4.500		50.00				Ar, 2.99	BOP & BW	Kutsuna, M., et al, 1993
	TEMP			100 00			_		9 9 9 9	1411

CONTINUOUS WAVE CO, LASER BEAM WELDING DATA FOR AL-ALLOYS APPENDIX A: (CONT'D):

MATERIAL: ALALLOYS
LASER BEANTYPE: CONTINUOUS WAVE

A-L-Cu-Mg-Zr	0.800 0.800 0.800 0.800 0.800 0.800 0.800 0.800 0.800	25.00 25.00 25.00 25.00 25.00 25.00 13.00 17.00 13.00 17.00 17.00 13.00 17.00 13.00 17.00 13.00 10.00 10.00 10.00 10.00 10.00 10.00 10.00 10.00 10.00 10.00 10.00	(kW/em-2) (kW/em-2) 1392.61 1591.55 596.83 1472.18	(kd/cm+2) (kd/cm+2) (67.3) 58.82 73.82	(e) (e)	Rate (m-3/hr)	Cherecterlative BOP & LW BOP & LW	
No. of the late Chief Ch	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	(mm/e) 10.00 22.1.40 7.00 7.00 7.00 25.00 42.33 42.33 25.00 13.00 17.00 17.00 17.00 6.35 6.35 6.35	(8W/6m*2) 1392.61 1591.55 596.83 994.72	(h.l.om·2) 67.31 58.82 73.82	9	Refe (m-3/h/)	Cherecteristies 8OP 5 LW BOP 5 LW	J.N. et al
CCZ 1.500 CCZ 1.800 CCZ 1.800 CCZ 2.000 CCZ 2.000 CCZ 1.300 CCZ 1	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	21.40 25.60 7.00 7.00 7.00 25.00 15.00 15.00 5.08 5.08 5.08 6.35 6.35	1392.61 1591.55 596.83 984.72 1472.18	67.31 58.82 73.82			80P & LW BOP & LW	J.N. 91 B.
1800 1800	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	25.60 7.00 7.00 7.00 7.00 10.00 10.00 13.00 17.00 17.00 17.00 17.00 5.08 5.08 5.08 6.35 6.35	1392.61 1591.55 596.83 984.72 1472.18	67.31 58.82 73.82			BOP & LW	
1.800 1.80	0 800 0 0 800 0 0 0 0	25.60 7.00 7.00 10.00 10.00 25.00 13.00 17.00 17.00 5.08 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82				Kamatu, J.N., et al, 1991
1,100 1,10	0 800 0 0 800 0 0 0 0	7.00 7.00 10.00 25.00 42.33 42.33 17.00 17.00 5.08 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82			BOP & LW	
COZ 2.000 COZ 4.300 COZ 4.1300 COZ 4.1300 COZ 6.1200 COZ 5.000 COZ 5.000 COZ 7.400 COZ 7.400 COZ 6.000	008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0	7.00 10.00 25.00 42.33 42.33 17.00 17.00 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82			BOP & LW	Kamalu, J.N., et al, 1991
002 1.300 002 1.300 002 EMO 1.300 002 EMO 1.300 002 2.000 002 7.400 002 6.000 002 6.000	0.85.0 0.	25.00 25.00 25.00 13.00 17.00 5.08 5.08 6.35 6.35	1392.61 1591.55 696.83 994.72 1472.18	67.31 58.82 73.82			BOP & LW	Kamalu, J.N., et al, 1991
CCZ 1.300 CCZ 1.300 CCZ 1.300 CCZ 1.300 CCZ 2.300 CCZ 2.000 CCZ 3.000 CCZ 3.00	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	25.00 42.33 254.00 13.00 17.00 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 994.72	67.31 58.82 73.82			BOP & LW	Kamalu, J.N., et al, 1991
	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	254.00 13.00 17.00 5.08 6.35 6.35 6.35	1392,61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82 123.03			MJ & GOB	Kamalu, J.N., et al, 1991
1.300 1.	008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0 008.0	254.00 13.00 17.00 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82 123.03			BOP & BW	Mollan, P.A., et al, 1988
COZ @ TEMO! 7.000 COZ @ TEMO! 8.000 COZ S.000 COZ C.000 C.	<u> </u>	13.00 17.00 5.08 5.08 6.35 6.35 6.35	1392.61 1591.55 596.83 994.72 1472.18	67.31 58.82 73.82 123.03			MB # dOB	Moltan, P.A., et al, 1988
COZ @ TEMO! 8.000 COZ S.000 COZ S.000 COZ S.000 COZ C.000 COZ S.000 COZ S.		17.00 5.08 5.08 7.44 7.45 8.35 8.35 8.35	1591.55 596.83 994.72 1472.18 994.72	58.82 73.82 123.03	0.061538462	Ar	BW	Blake, A., et al, 1985
COZ 5.000 COZ 7.400 COZ 7.400 COZ 7.400 COZ 6.000		5.08 6.35 7.47 8.35 8.35 8.35	596.83 994.72 1472.18 994.72	73.82	0.047058824	Ϋ́	BW	Blake, A., et al, 1985
COZ		5.08 6.35 7.47 8.35 8.35 8.35	994.72 1472.18 994.72	123.03	0.157480315		æ	Blake, A., 1962
COZ (5.000		6.35 6.35 6.35	1472.18		0.157480315		Q t	Blake, A., 1962
COZ 6.000		6.85 7 8.85 8.85 8.85 8.85	994.72	182.09	0.157480315		Đ.	Blake, A., 1962
COZ		6.35 6.35 8.35		73.79	0.094451004		83	Blake, A., 1982
COZ 6.000		6.35 6.35	786.77	78.74	0.125984252	£	608	Blake, A., 1962
COZ 6.000	$\Box \Box$	6.35	994.72	98.43	0.125984252	#	908	Blake, A., 1982
COZ 6.000 COZ 7.000 COZ 6.000	$oldsymbol{ol}}}}}}}}}}}}}}}}}$		1193.66	118.11	0.125984252	£	83	Blake, A., 1982
COZ 6.000	L	6.35	1591.55	157.48	0.125984252	£	8	Blake, A., 1982
COZ 6.000		10.58	795.77	47.26	0.075614367	事	82	Blake, A., 1962
COZ 6.000	0.800	10.58	994.72	59.07	0.075614367	2	208	Blake, A., 1982
COZ 6.000 COZ 6.000 COZ 8.000 COZ 6.000 COZ 8.000 COZ 8.000 COZ 6.000 COZ 6.000	L	10.58	1193.66	70.89	0.075614367	Ŧ	Đ.	Blake, A., 1962
CO2 8.000 CO2 8.000 CO2 8.000 CO2 8.000 CO2 8.000 CO2 9.000	0.800	10.58	1591.55	94.52	0.075614367	£	£	Blake, A., 1962
CO2 8.000 CO2 6.000 CO2 8.000 CO2 8.000 CO2 8.000	0.800	19.05	1193.66	39.37	0.041994751	£	8	Blake, A., 1962
CO2 8.000 CO2 8.000 CO2 6.000 CO2 8.000 CO2 9.000	0.800	19.08	1591.55	52.49	0.041994751	4	808	Blake, A., 1982
CO2 8.000 CO2 6.000 CO2 002 8.000 CO2 002 8.000	Ш	12.70	1193.66	59.06	0.062992126	ž	£	₹
COZ 6.000 COZ 8.000 COZ 6 TEMO! 5.000 COZ 6 TEMO! 7.000	0.800	12.70	1591.55	78.74	0.062992126	皇	æ	۲
COZ @ TEM01 5.000	Ш	31.75	1193.66	23.62	0.02519685	2	å	
CO2 @ TEM01 5.000	0.800	31.75	1591.55	31.50	0.02519685		8	Blake, A., 1982
CO2 @ TEMO1 7.000	0.800	6.35	994.72	98.43	0.125984252	£	80	Blake, A., 1982
COC & LANGE TO COC	0.800	6.35	1392.61	137.80	0.125984252	2	8	Blake, A., 1982
2.000	0.800	8.47	994.72	73.79	0.094451004	2	8	Blake, A., 1982
7.000	0.800	8.47	1392.61	103.31	0.094451004	2	8	4
_	0.800		200.00				6 2	Blake, A., 1982
_	0.800		400.00				8	₹
-	0.800		600.00				8	Blake, A., 1982
CO2 @ TEM01 4.000	0.800		800.00				å	Blake, A., 1982
CO2 @ TEM01 5.000	0.800		1000.00				8	∢
CO2 @ TEMO1 6.000	0.800		1200.00				8	Blake, A., 1982

BOP (Bead-on-plate), BW (Butt Weld), FP (Full Penetration), IBW (I-Butt Weld), LW (Lap Weld)

APPENDIX B: PULSED LASER BEAM WELDING DATA FOR AL-ALLOYS

MATERIAL: ALALOYS
LASER BEAN TYPE: PLLSED

Mederhei	Sayl seal	Arestage	1	Process) demand	Dectile.	1		7	Interection	Opposition	Y•*	
Sevelatoritors	4	Ì	Dinmette		Dens 1	Energy .	Ducanton	Fragones	2 2 2	=	į		Pe t.
A Geometry	1	1	_	(0/WW)	(hW/em-2)	(14/em-1)			2	101	See ja-34.	Cherecteristics	
A5052P-H32	8	- 400	Т	6.67				200			۸۲		Masumato, et al., 1992
l=1mm to 6mm	8	1.400		10.00				200			Ar .		Masumato, et al, 1992
	8	99-		16.67				200			Ar		Masumato, et el, 1992
	88	1.400		33.33				200			A,		Masumato, et al, 1992
	8	1.400		20.00				200			Ar		Masumato, et al, 1992
6061, 1×3mm	ΟX	2.000		8.8				110			Ar, 15	BW & LW	Yoshikawa, M., et al, 1995
	λγ	2.000		90.9				110			ł	BW & LW	Yoshikawa, M., et al. 1995
	YAG	2.000		8.8				110			N2	BW & LW	Yoshikawa, M., et al, 1995
	OXX	2.000		8.67				110			1	BW & LW	Yoshikawa, M., et al, 1995
	YAG	2.000		6.87				110			M2	BW & LW	Yoshikawa, M., et al., 1995
	YAG	2.000		25.00				110			Ar, 1.5	BW & LW	Yoshikawa, M., et al, 1995
5063, 1-2mm	N.Y	2.000		25.00			2.20	110			f	BW & LW	9
	AAG	2.000		25.00				110			ž	BW&LW	Yoshikawa, M., et al, 1995
	DYA	2.000		16.67				110			Ar, 1.5	BW&LW	Yoshikawa, M., et al, 1995
	SAY.	2.000		16.67				110			1	BW & LW	Yoshikawa, M., et al, 1995
	SAY.	2.000		16.67				110			N2	BW & LW	Yoshikawa, M., et al, 1995
6061. 1=2mm	NAG.	2.000		25.00				110			Ar, 1.5	BW & LW	Yoshibawa, M., et al., 1995
	YAG	2.000		25.00				110			£	BW&LW	Yoshakawa, M., et al, 1995
	¥¥0	2.000		25.00				110			N2	BW&LW	Yoshikawa, M., et al, 1995
5083, 1=3mm	YAG	2.000		6.33				110			Ŧ	BW & LW	Yoshikawa, M., et al, 1995
6061. (=3mm	λ¥G	2.000		4.17				110			¥6, №	BW&LW	Yoshakawa, M., et al, 1995
6061	N d: YAG	0.020	0.400				0.007	-			Ar		Coyle, R.J., et al, 1967
413	Nd:YAG	0.020	0.400				0.007	-			Ą		Coyle, R.J., et al., 1987
4047	NAYAG	0.020	0.400				0.007	-			₹		Coyle, R.J., et al. 1987
7.1mm dla. x 0.3mm]				
5000 series	NA:YAG	3.000		38.33								X	Riches, S.T., 1993
9000 series	NEYAG	3.000		33.33								à	Riches, S.T., 1998
(=1.25mm		_						1			1		All and A second
AMAGE	COZ @ TEMOO	- 504.	9. 9.		17825.35			20000			OZNZH		Michigan, A., 91 M., rusyes
23.											7		× 6 4 6 4
				_					_	_	1.0 0 2.1		** P 0: 0'

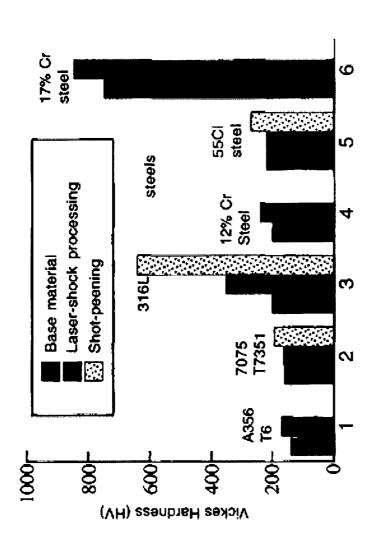
SW (Bust Weld), LW (Lep We

CONTINUOUS WAVE CO2 LASER BEAM WELDING DATA FOR TITANIUM ALLOYS APPENDIX C:

Material	Laser Type	Average	E 28 G	Process	Power	Specific	Interaction	Shleiding	PIOM			
	•	Power	Dieneter	D-0-d	Density	Energy	÷ E 2	Cas & Flow		.	Aef.	
Specification	Deam Mode	(kW)	(E E)	(mm/s)	(kW/cm^2)	(kJ/cm^2)	3	Rate (m-3/hr)	Cheracteristics			
TI-6AL-4V.	800	1.000	0.250	23.00	2037.18	17.39	0.010869565	He, Ar	BW.	Mazumder, J., et al,		1982
AMS 4911-Te10 BSS	88	1.000	0.250	34.00	2037.18	11.78	0.007352941	He, Ar	BW	Mazumder, J., et al,	J., et al,	1982
-1 mm	88	1.000	0.250	15.00	2037.18	26.67	0.016666667	He, Ar	BW	Mazumder, J., et al, 1982	J., et al,	1982
1-1mm	88	1.000	0.250	11.00	2037.18	36.36	0.022727273	He, Ar	BW	Mazumder, J., et al,	J., et al,	1982
1=0.89mm	88	1.480	0.250	60.00	3015.03	9.87	0.004166667	He, Ar	BW.	Mazumder, J., et al,	J., et al,	1982
1=0.89mm	800	1.480	0.250	50.00	3015.03	11.84	0.005	He, Ar	BW	Mazumder, J., et al,		1982
t=0.89mm	200	1.480	0.250	23.00	3015.03	25.74	0.010869565	He, Ar	BW	Mazumder, J., et al, 1982	J., et al,	1982
1=0.81mm	88	1.500	0.250	60.00	3055.77	10.00	0.004166667	He, Ar	BW	Mazumder, J., et al,	J., et al,	1982
t=0.81mm	200	1.500	0.250	50.00	3055.77	12.00	0.005	He, Ar	BW	Mazumder, J., et al,		1982
t=0.81mm	800	1.500	0.250	23.00	3055.77	26.09	0.010869565	He, Ar	BW	Mazumder, J., et al,		1982
1=2.03mm	200	1.500	0.250	7.00	3055.77	85.71	0.035714286	He, Ar	BW	Mazumder, J., et al, 1982	J. et al,	1982
t=2.03mm	8	1.500	0.250	9.00	3055.77	66.67	0.02777778	He, Ar	BW	Mazumder, J., et al,	J., et al,	1982
t=2.03mm	8	1.500	0.250	13.00	3055.77	46.15	0.019230769	He, Ar	BW	Mazumder, J., et al,	J., et al,	1982
1188	8	1.110	0.250	6.00	2261.27	74.00	0.041666667	He, Ar	BW	Mazumder, J., et al, 1982	J., et al,	1982
t-1	8	1.500	0.250	23.00	3055.77	26.09	0.010869565	He, Ar	BW	Mazumder, J., et al, 1982	J., et Bl,	1982
1133	800	1.540	0.250	50.00	3137.26	12.32	0.005	He, Ar	₩	Mazumder, J.,	et si,	1982
t=1mm	8	1.540	0.250	60.00	3137.26	10.27	0.004166667	He, Ar	BW	Mazumder, J., et al,	J., et al.	1982
BW (But Weld)												

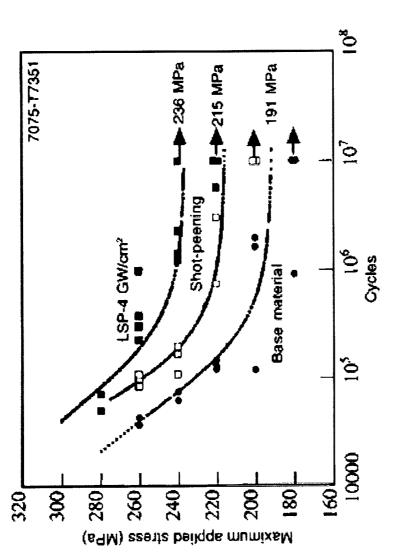
148

LASER SHOCK PROCESSING DATA OF ALUMINUM ALLOYS APPENDIX D:



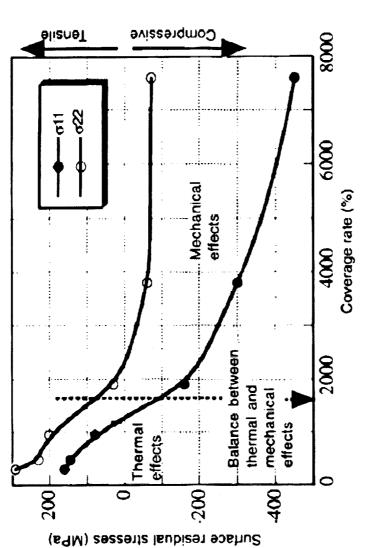
cm², water confined regime, ambient atmosphere) - Comparison with shot-peening.27.34.39 Maximum surface hardening induced in a typical LSP configuration (20-30 ns, 5-10 GW/

LASER SHOCK PROCESSING DATA OF ALUMINUM ALLOYS APPENDIX D (CONT'D):



shock processing surface treatments (notched samples, bending-fatigue at 30 Hz - R = $0.1)^{28}$ S-N curves on a 7075-T7351 aluminium alloy, influence of shot-peening and laser-

LASER SHOCK PROCESSING DATA OF ALUMINUM ALLOYS APPENDIX D (CONT'D):



LSP without protective coating: surface stress dependence on coverage factor (AISI304 steel -5 ns pulses- 0.75 mm spots).

APPENDIX E: TORNADO ND:YAG LASER SPECIFICATIONS

	eneral Characteristics
Model	TN50-106Q
Wavelength	1064 nm
Average Power	>35 W @ 10 kHz
Pulse Width (nominal)	120 ns @ 10 kHz
Pulse Eenergy, 5 kHz	>5 mJ
Pulse Energy, 10 kHz	>3.5 mJ
Pulse Duration, 1 kHz	>80ns
Pulse Duration, 5 kHz	90 ns, nominal
Pulse Duration, 10 kHz	110 ns, nominal
Repetition Rate Range	Single shot to 80 kHz
Spatial Mode	Flat-Top (Multi Mode Beam) M² < 14 at 10 kHz
Polarization	Unpolarized Beam

Beam Specifications

\mathbf{M}^2	< 15
Polarization	Random
Beam Diameter	
(at waist)	1.9 mm (nominal)
Beam Diameter	2.0 mm @ OC
@ 10 kHz	
Beam	< 10% difference in any
Ellipticity	orthogonal axes
@ 10 kHz	or thogonal axes
Beam	$4.5 \text{ mrad } (M^2 = 14)$
Divergence	4.5 mr au (W 14)
Boresight:	0.1 mm position of output
Stability	3
Beam Pointing	<20 ~rad per oC
Pulse-to-pulse	Energy (5-10 kHz) <2% rms
Stability	Peak Power (5-10 kHz): 3.5% rms
Long-term	3% non-cumulative, over 8 hours
Stability:	Environmental Characteristics
Warm-up	<10 minutes
Time	(cold start to 90% of full power)
Temperature	18°-35° C
Range	10 -33 C

Altitude	0-3,000 meters
II	8-95%, non-condensing
Humidity	Non-Operating Conditions
Temperature	0-50°C

Physical Characteristics

ac, 50/60 Hz
(typical)
CFM
(13 Ibs)
(30.5 Ibs)
(86 Ibs)

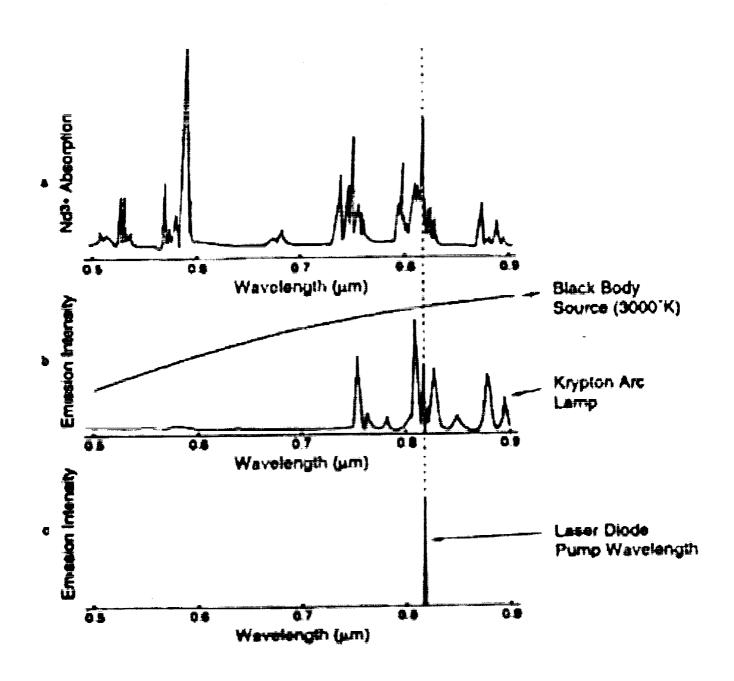
Chiller Requirements

Electrical Requirements	200- 240 Vac, 10 A, 50 or 60 Hz
Temperature	20°C
Flow Rate	(0.5 Gal/min)
Pressure	344 kPa (50 psi)
Cooling Capacity	> 1 kW
Coolant	Deionized Water

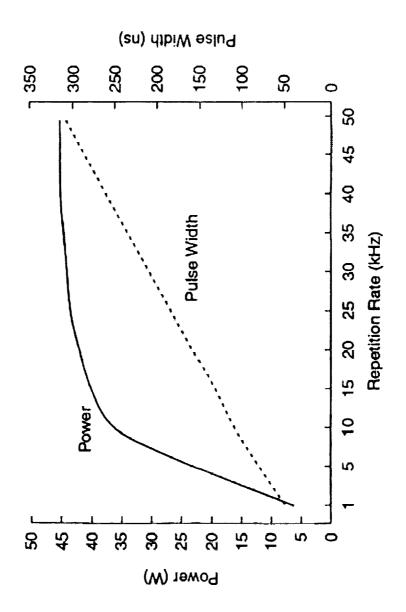
Diode Module Specifications

Lifetime	5250 hours
Field Replaceable	(8760 hrs is 1 year)

APPENDIX E (CONT'D): TORNADO ND:YAG LASER SPECIFICATIONS



APPENDIX E (CONT'D): TORNADO ND:YAG LASER SPECIFICATIONS



GLOSSARY OF TERMS FOR LASER CHARACTERISTICS APPENDIX F:

= Pulse Energy (Joules) x Pulse Repetition Rate (Hz) Pa = ER AVERAGE POWER

 Peak Power Pulse (Watts) x Pulsewidth (Sec.) x Pulse Repetition Rate (Hz Pa = P₂TR AVERAGE POWER (Watts)

(Beam Diameter (MM) x Beam Divergence (mr)) Power (Watts) ۵. BRIGHTNESS

(00)

= Peak Power Pulse (Watts) x Pulsewidth (Sec.) E= P_eT

(Joules)

PULSE ENERGY

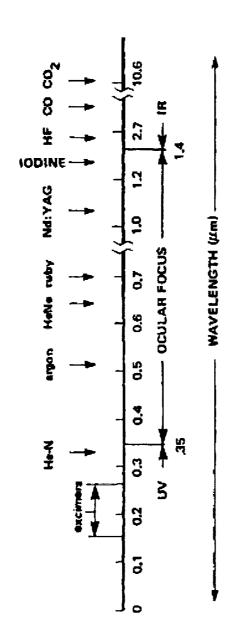
Intensity of light after passing through filter LIGHT TRANSMISSION

Intensity of light = -1 Log (Light transmission) $T\lambda = 10^{10}$ 11 OPTICAL DENSITY

= Diffraction Limited spot + Spherical Aberration d. = 1.27 MM

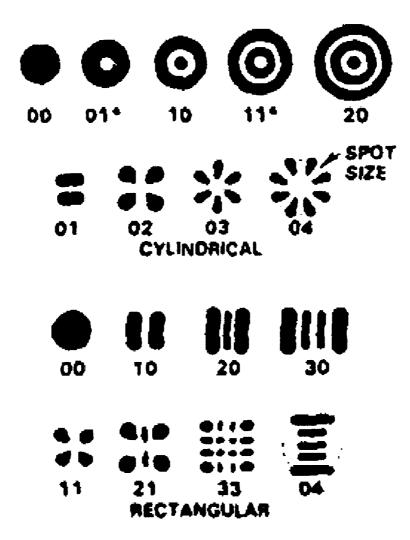
SPOT SIZE

APPENDIX F (CONT'D): GLOSSARY OF TERMS FOR LASER CHARACTERISTICS



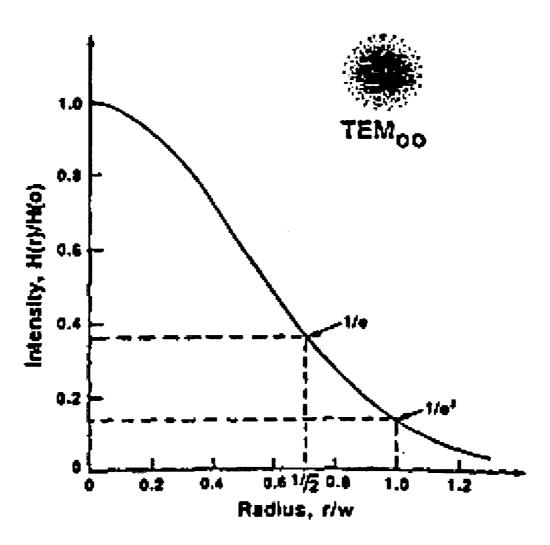
Wavelength spectrum of typical lasers

APPENDIX F (CONT'D): GLOSSARY OF TERMS FOR LASER CHARACTERISTICS



Transverse Electromagnetic Modes TEM_{mn}

APPENDIX F (CONT'D): GLOSSARY OF TERMS FOR LASER CHARACTERISTICS



Gaussian Distribution of TEM...

APPENDIX G: TYPICAL MELT ENERGY CALCULATIONS FOR TITANIUM AND ALUMINUM ALLOYS

Alloy	Comp	onen	t Per	centa	ge by	y We	ight
Material	Ti	Al	v	Cu	Li	Mn	TOTAL S
Ti-6Al-4V	0.900	0.060	0.040	0	0	0	1.000
Al 2090	0	0.950	0	0.027	0.023	0	1.000
Al 2219	0	0.930	0	0.067	0	0.003	1.000

Empirical and Calculated Alloy Thermal Property Values										
Material	Density (kg/m³)	Melting Point (°C)	Heat Capacity (J/kg-°C)	Thermal Conductivit y (W/m-K)	Heat of Fusion (J/kg)	Melt Energy (J/mm³)				
Ti-6Al-4V	4430	1660	526.30	6.7	427653.20 ^a	5.7182 ^b				
Al 2090	2590	650	1203.00	88	376235.60 ^a	2.9374 ^b				
Al 2219	2840	643	864.00	120	374351.60°	2.5919 ^b				

Values in blue obtained from www.matweb.com

Values in green obtained from Alcoa Green Letter, #226, 1988

Values in red calculated from obtained values

a: Weighed average calculated using primary alloying elements

b: Calculated by: $E_M = 1_{mm}^3 \times \rho \times \left[(C_P \times \Delta T) + \Delta H_f \right]$

Where: $T = (Melting Point - 20^{\circ}C)$

C_P= Specific Heat

ΔH_c=Enthalpy of Formation

ρ=Density

APPENDIX H:
COMPARISON CHART OF VARIOUS LASERS

Electrical Requirements	208V, 3PH, 20A	460V, 3PH, 100A	208V, 3PH 208V, 3PH	220V, 1PH, 20A	440V, 3PH, 100A 208V, 3PH, 192A	480V, 3PH		208V, 3PH (excluding chiller)
Approx. Cost (US\$)	32, 700 57, 500	70, 000	47, 000 95, 000	38, 000 108, 000	106, 000	39, 000	125, 000	110000
Output Power (W)	50 100	500	500 1000	500 100	700	400	1000	500
Beam Mode	pulsed/	cw	cw	cw pulsed/cw	cw	pulsed	cw	cw
Pumping Medium	diode diode	lamp	lamp lamp	lamp diode	dmel	lamp	direct diode laser	lamp lamp
Wavelength (µm)	1064	1064	1064	1064 532	1064	1064	810	1064
Manufacturer & Model	Spectra Physics Tornado- TN50-106Q (new) TN100-106Q (new)	GSI Lumonics JK500 (used)	USLaser 406-1 (new) 406-2 (new)	Lee Laser 8250M/500 (new) LDP-200MQG (new)	Trump HL703D (new)	Raytheon SS550 (used)	Nuvonyx* ISL-1000L (new)	Roffin-Sinar RSY-500 RSY-1000

* Typical Wall Plug Efficiency: CO2: 12%, Lamp Pumped Nd:YAG: 1-2%, Diode Pumped ND:YAG: 8-12%, Direct Diode Laser: 40%